

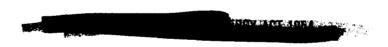
AEROJET GENERAL

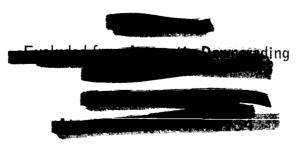
REPORT NO. RN-S-0050
TO
AEC-NASA SPACE NUCLEAR PROPULSION OFFICE
RADIATION ANALYSIS OF MECHANICAL IRRADIATION TESTS
901 THROUGH 904 (U)

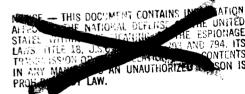


ROCKET ENGINE OPERATIONS - NUCLEAR

NERVA PROGRAM CONTRACT SNP-1 JANUARY 1964







A E R O J E T - G E N E R A L CORPORATION

A SUBSIDIARY OF THE GENERAL TIRE & RUBBER COMPAN'

... LOIDANG PAGE IN TILL ON A HILLO

RN-S-0050

ABSTRACT

This report contains the results of radiation analyses performed in support of Mechanical Irradiation Tests 901, 902, 903 and 904. Comparisons are made between the calculated test environment and the comparable NERVA flight environment. Comparisons are also made with experimental results in those cases where experimental data were available.

Fast neutron flux and gamma ray dose-rates and integrated doses are reported, as well as the neutron-to-gamma number flux ratio for each component as it appears in NERVA and the GD/FW ASTR test. Gamma heating rates are also presented in the same context.

This report was prepared by E. A. Warman and P. H. Fogg of the Nuclear Analysis Department. Coordination with GD/FW and the Liquid Rocket Plant was provided by the Radiation Effects Department at REON.

/ D. Stinnett

Ellottine

MERVA Technical Systems Manager

REOI

CONTENTS

Abstr	act	, , , , , , , , , , , , , , , , , , , 			
ı.					
II.	OBJECTIVES				
III.					
	A. MIT 901				
	B. MIT 902				
	C. MIT 903				
	D. MIT 904				
IV.	RADIATION ANALYSIS				
	A. Gamma Radiation Analysis				
	B. Neutron Analysis				
V.	CALCULATED RADIATION ENVIRONMENT				
	A. MIT 901				
	B. MIT 902				
	C. MIT 903				
	D. MIT 904				
VI.	COMPARISONS WITH EXPERIMENTAL RESULTS				
REFER	ENCES				
<u>Table</u>					
1	Gamma Ray Source Strength and Spectrum				
2	Radiation Environment - MIT 901				
3	Comparison of NERVA and ASTR-901 Test Environments				
<u> 1</u>	Radiation Environment and Heating Rates				

CONTENTS (cont.)

Table		Pa
5	Comparison of NERVA and ASTR Environment at the Turbopump Bearing Locations	
6	Comparison of Calculated and Experimental Data	
Figur	<u>e</u>	
1	Modified ASTR (10 Mev)	_
2	MIT Test Pallet with Shroud in Place	
3	MIT Test Pallet with Shroud Removed	_
14	Mechanical Component Locations in NERVA	_ :
5	TSOV Location in MIT 901 Test	_ :
6	MIT 901 Bearing Tester (Photograph)	_ :
7	MIT 901 Bearing Tester Assembly	_ ′
8	Refined Bearing Tester Assembly	
9	Pulse Height Spectrum of Gamma Rays from the 10 Mev ASTR	
10	Unperturbed Gamma Ray Dose Rates in ASTR Test Cell	_
11	Analytical ASTR Neutron Spectrum	-
12	Unperturbed Fast Neutron Fluxes in ASTR Test Cell	
13	Analytical Model - TSOV	_
14	Analytical Model - TPCV	
15	Analytical Model - Bearing Tester	·
16	Dose Point Locations for MIT 901	

I INTRODUCTION

LALUEDING PAGE THE CONTROL FILMED.

RN-S-0050

I. INTRODUCTION

This report describes the results of a radiation analysis of the environment to which mechanical components of the NERVA engine are exposed at the General Dynamics Aerospace Systems Test Reactor (ASTR). Mechanical irradiation tests 901 through 904 are included in this analysis. (See References 1 through 4.)

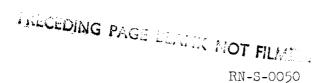
Neutron and gamma radiation intensities and heating rates were calculated with the aid of the QAD point kernel computer code (Ref. 5). Sources of radiation used in these calculations are discussed along with the analytical techniques used to describe the test environment. In all cases, the comparable NERVA flight environment is reported in conjunction with the ASTR test levels.

Comparisons are made between the analytical results and measured data in those instances where the tests have already been completed.

This report supersedes, and replaces all data reported in, Reference 1. The earlier report represented an initial attempt to pre-analyze the 90l radiation environment. When the measured spectral data was made available by GD/FW in Reference (6) the improved treatment reported herein became possible. Future reports in the MIT series will discuss improved analytical approaches as they relate to experimentation.

ALCEDING PAGE LLATK NOT FILMED

II OBJECTIVES



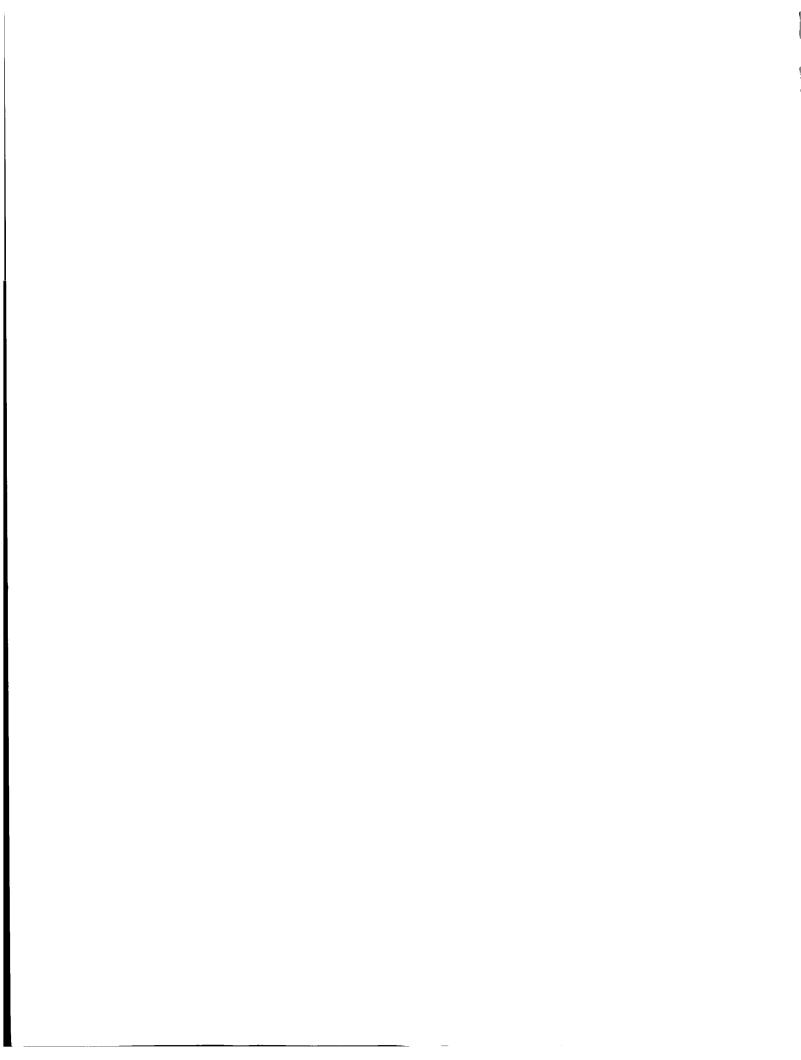
II. OBJECTIVES

The major purpose of this report is to document a comparison between NERVA radiation environment and the test environment to which NERVA mechanical components are subjected at the ASTR facility. Both the NERVA and ASTR data are analytical values. However the ASTR analyses are based on a normalization to experimental data in the empty test cell (Reference 6).

Comparisons are also made with measured data for tests which have already been completed. The objective in this analysis as a function of experiment comparison is:

- A. To achieve greater assurance of the accuracy of analytical techniques.
- B. To allow interpolation to any number of dosimeter points in the analytical model to augment the experimental data points.

In general, experimental data is not and will not be available at the most important locations in any given test, for example, the bearings in a bearing tester. It is intended that the data produced by this analysis should serve to fill in this important gap in the test results.



I RECEDING PAGE BLANK NOT FILMES:

III DESCRIPTION OF TESTS

THE PAGE DESCRIPTION OF PILMERY

RN-S-0050

III. DESCRIPTION OF TESTS

The mechanical irradiation tests were conducted at the GD/FW Aerospace Systems Test Reactor (ASTR). A 10 Mw (thermal) reactor capable of producing radiation levels comparable with those to which components will be exposed in a NERVA flight environment. Since the higher power NERVA engine includes an internal radiation shield, the shielded engine component radiation levels may be compared to the lower power but unshielded ASTR values. Figure 1 shows the test cell configuration at ASTR for this series of tests.

The particular components included in each test are described, in the following pages, together with a general discussion of the nuclear and non-nuclear aspects of the test. Greater detail concerning the nuclear aspects of these components is contained in the Radiation Effects Data Book, Reference (7). Further details of the non-nuclear aspects are contained in the relevant test requests, test plans, and specifications contained in Radiation Effects Testing Data Book, Reference (8).

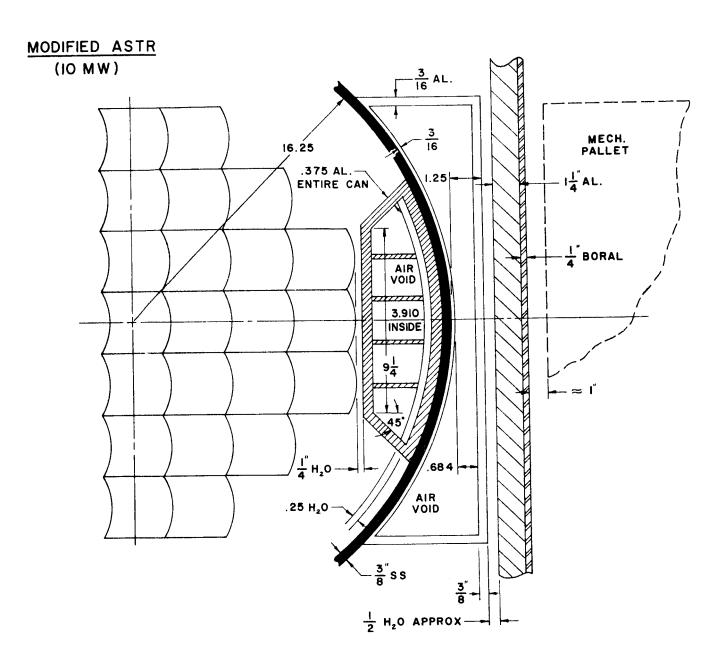


Figure 1
Modified ASTR (10 Mw)

A. MIT-901

Mechanical Irradiation Test 901 includes the following four individual tests. Figure 2 is a photograph of the MIT-901 test pallet with the shroud in place. Figure 3 shows the pallet with the shroud removed.

1. Radiation Effects Test 1/L001

Radiation Effects Test 1/L001 involves Configuration 1 of the Tank Shutoff Valve (TSOV).

Figure 4 shows the TSOV in the NERVA context. This valve is a ${
m CH}_2$ -operated poppet valve designed for completely open or completely closed operation to control ${
m IH}_2$ propellant flow from the propellant tank into the propellant feed line.

In this test the TSOV was immersed in LH_2 with the fluid present on both sides of the valve. The liquid flow rate when the valve is open will be relatively small, approximately 0.05 lb/sec. During irradiation the valve was periodically cycled from closed, to open, to closed.

Figure 5 shows the TSOV in the context of Test 901. The position transducer with its mechanical linkage is visible at the left side of the valve, the actuating gas line is on the right, and the LH₂ reservoir is below. The static flange and seal tester with its multiple Marman clamp configuration, and the bearing tester support are in the foreground.

2. Radiation Effects Test 3/LOO1

Radiation Effects Test 3/LOO1 involves the Turbopump Bearing Assembly, shown in the NERVA context in Figure 4.

This assembly includes the turbopump shaft with its bearing sets and bearing spacers, the housing, and the shaft RPM transducer. An actuating electric motor is provided to drive the shaft from the turbine end at approximately 21,000 rpm. A closure at the pump end is also provided.

No attempt was made to approximate the axial loads of the NERVA application. During the test LH $_{\odot}$ flowed through the assembly at about 0.1 lb/sec.

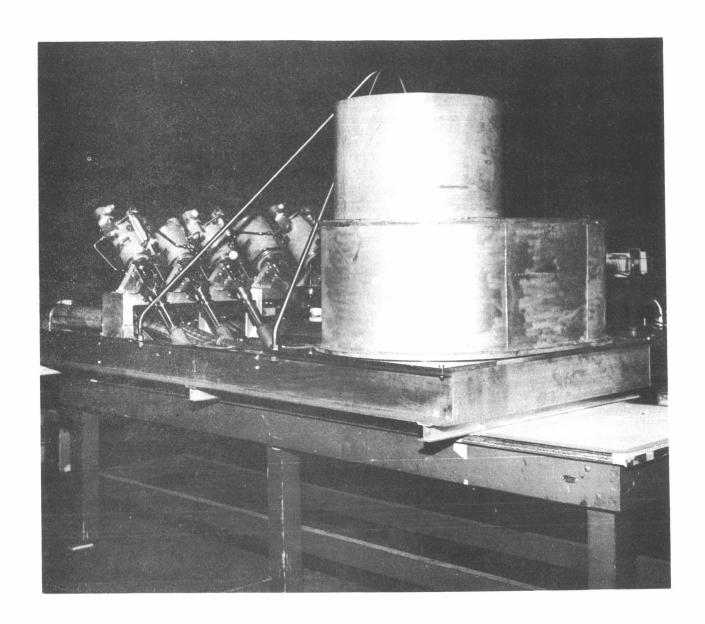


Figure 2
MIT Test Pallet With Shroud in Phase

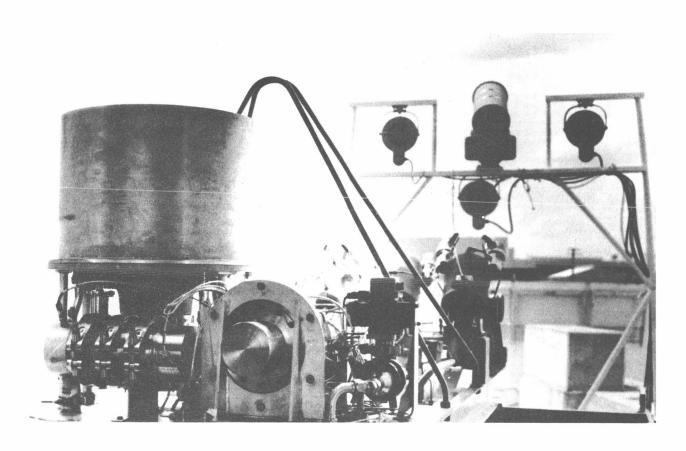


Figure 3
MIT Test Pallet with Shroud Removed

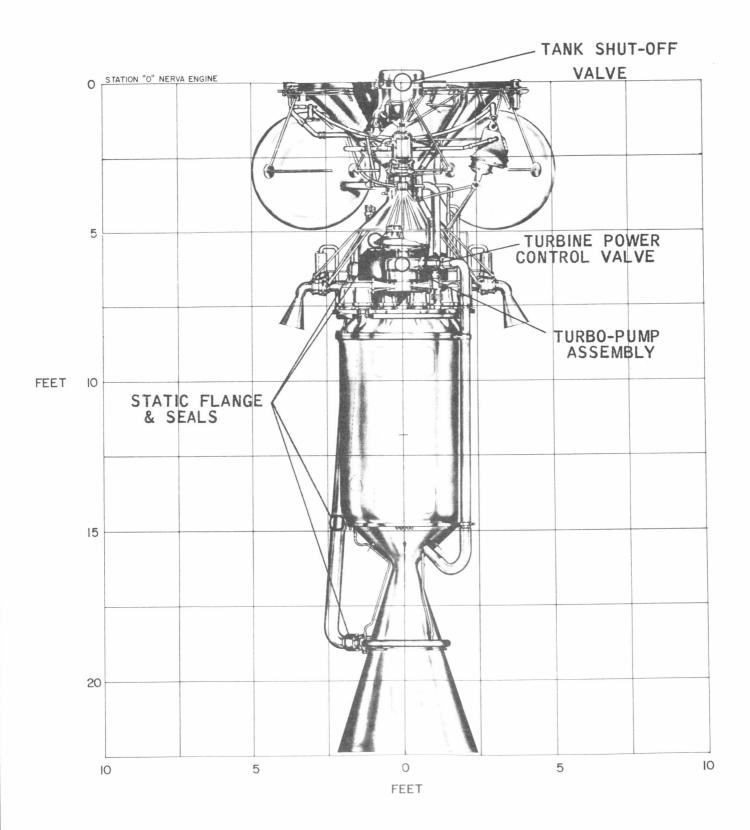


Figure 4

Mechanical Component Locations in NERVA

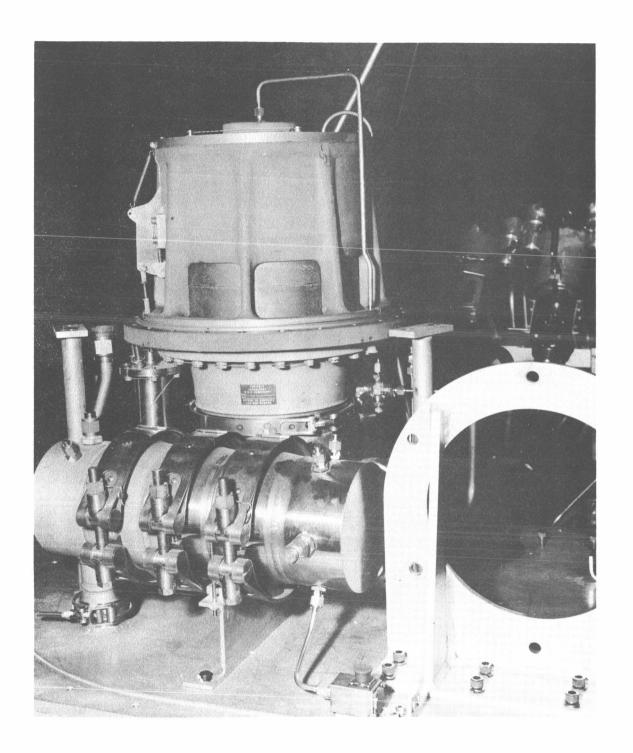


Figure 5
TSOV Location in MIT 901 Test

Figures 6 and 7 show the bearing assembly tester. The section from the middle to the right end is the electric motor drive; the turbine-end bearing-temperature transducer connections are seen in the center of the picture. The connection to the left of center is the rpm transducer connection. The support flange is seen at the left.

3. Radiation Effects Test 8/L001

Radiation Effects Test 8/L001 involves static flanges and metal-to-metal seal combinations. Typical examples of this are shown on the NERVA engine in Figure 4. The test configuration includes three seal combinations; aluminum-to-aluminum, aluminum-to-steel, steel-to-steel. The end covers are secured by Marman clamps to form an enclosure.

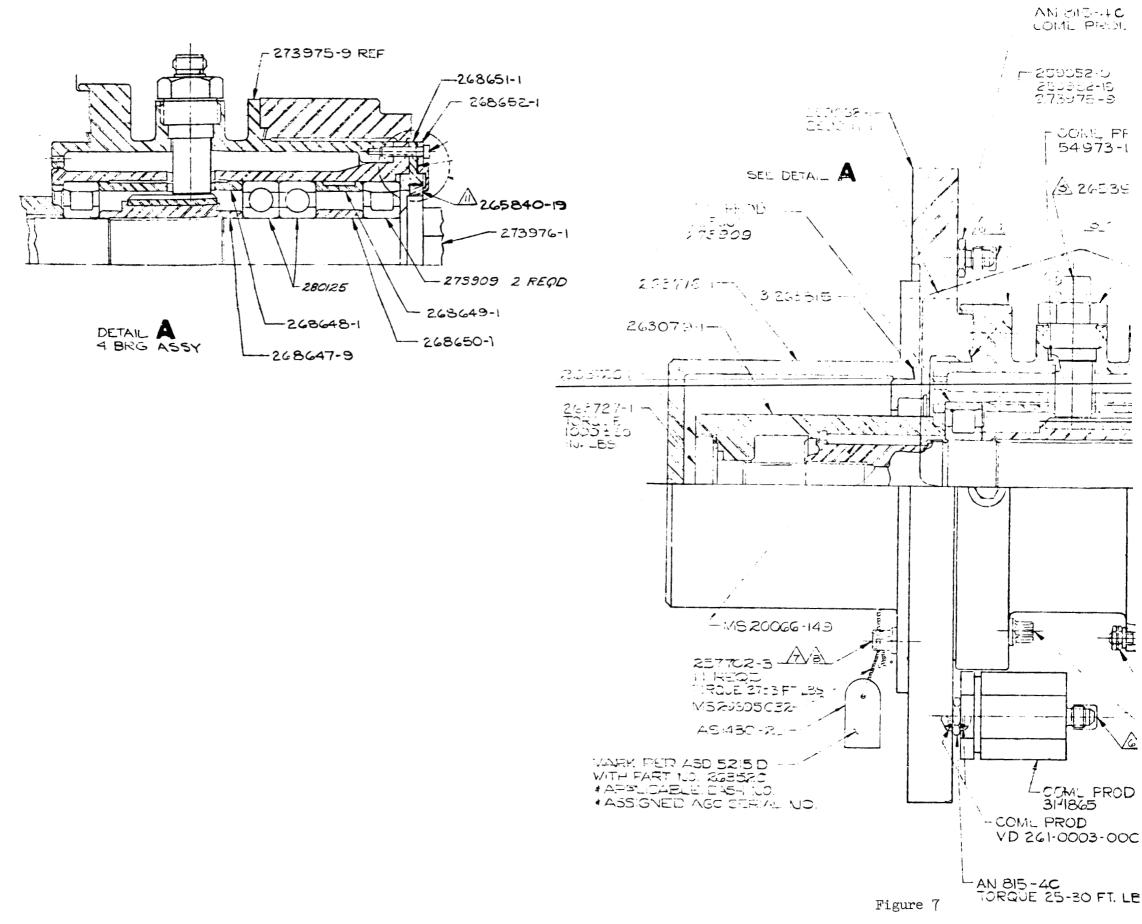
Reference is again made to Figure 3. The seal and flange test configuration is mounted on the test pallet with the aluminum seals to the left, and the steel flanges to the right. The LH₂ pressure transducer is located in the lower center foreground.

4. Radiation Effects Test 4/LOOL

Radiation Effects Test 4/L001 involves the Turbine Power Control Valve (TPCV) which is located on the NERVA engine in Figure 4. The TPCV is a butterfly-type valve designed to regulate the flow of high temperature GH_{2} to the turbopump turbine, thus controlling the turbine power. The loss of the valve or knowledge of its position during NERVA operation would be most serious for engine operation, making imperative the design of a reliable valve.

The test configuration includes a pneumatic linear actuator which will position the valve blade during the test to regulate the flow of ambient temperature GH_2 . The test actuator is not intended for NERVA application. The operational actuator is being developed under a different program, and will be mated to the TPCV in a later radiation effects test.

Figure 3 also shows the TPCV with its actuator, and other test components, mounted in their relative positions on the test pallet. The light colored box on top of the actuator is the position sensing transducer. The valves



MIT 901 Bearing Tester Assembly

DUI FRANT Q

James Aug.

OD 2

5.

259954-1 \circ TO STATE NULBE 268610-3 ~ T2. 0754-9 572-9 -2693804 - 259972-1 256209-1 2 RE /: -259955-1,-3*0*R-5 AN 316 C-4 TORQUE 300 =25 W. LBS -- COML PROD <u>ŝ</u>coml PROD FIC - 428-1 257.83-3 2 REQU 257.83-7 10 REQD M620995032 IZ RECID TORGUE 115 ± 5 IN. LBS 7331-1 ď COML PROD NO. 5660 280125 TORQUE 11515 IN. LBS. -257702-5/A 8 REQD. MS 20995C32 TORQUE 27:3 FT. LB5. 4

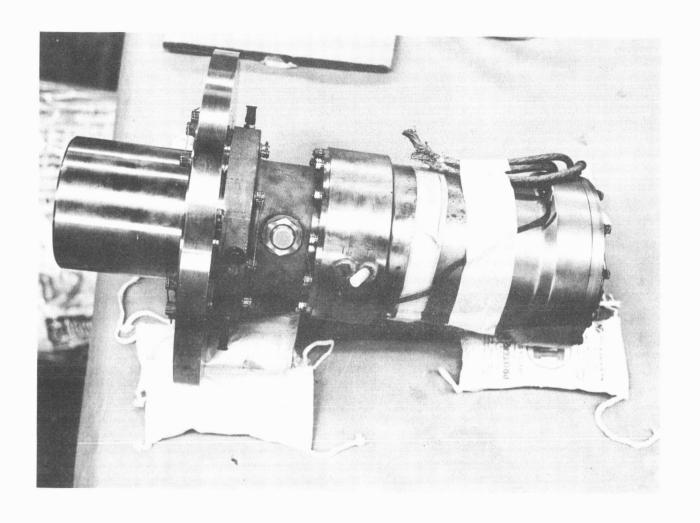


Figure 6
MIT 901 Bearing Tester (Photograph)

RN-S-0050

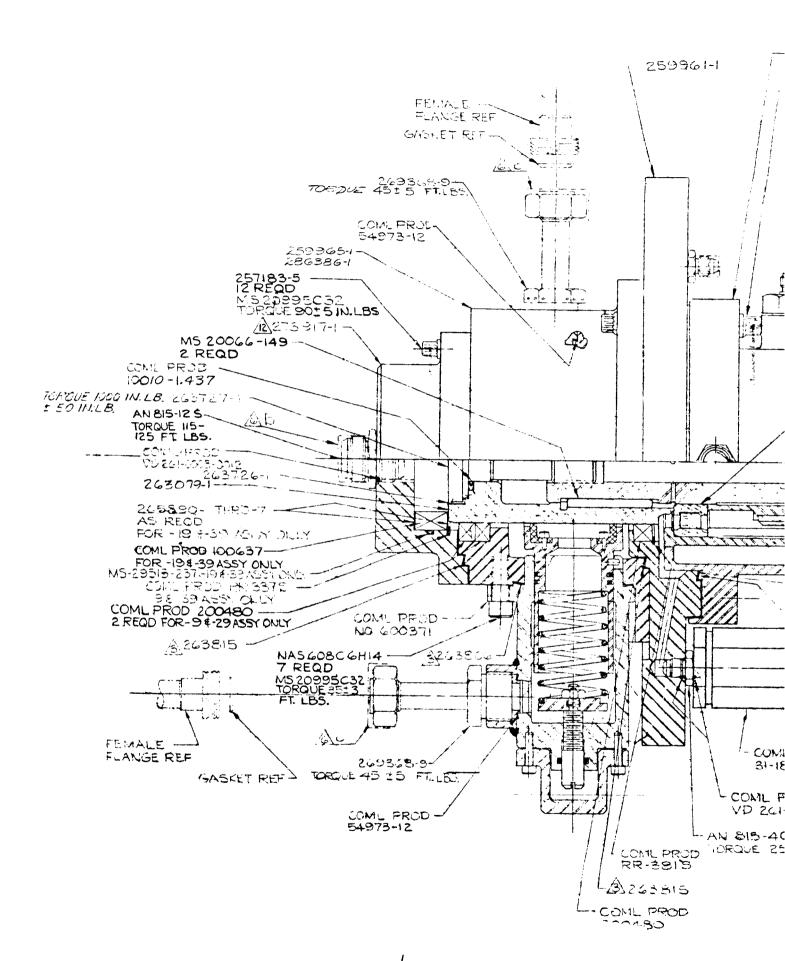
which regulate the environmental gases and liquids can be seen in the background, as can some of the necessary piping. The TSOV is enclosed in a cryostat to contain the $\rm LH_2$ bath. The remainder of the components are enclosed by a cover (as shown by Figure 1), which contains an inert atmosphere of helium to prevent an explosion of leading $\rm GH_2$. Test duration, at full reactor power of 10 megawatts, was 327 seconds.

B. MIT-902

The test hardware for the MIT-902 test consisted of a modified MIT-901 bearing tester enclosed by itself on the MIT-901 pallet. The MIT-901 bearing tester was modified to permit radial and axial loading of the turbopump shaft. The modifications added an appreciable amount of stainless steel in the end of the tester towards the reactor. This has the effect of causing a reduction (from the results of the MIT-901 test) in gamma dose and heating rates and in the neutron flux levels at the bearings located behind the added material. Figure 8 is an engineering drawing showing the modifications and the Armalon layed bearing configuration tested. Test duration at reactor full power was 300 sec.

C. MIT-903

MIT-903 test used the MIT-902 bearing tester and pallet unmodified. Armalon caged bearings were again tested, however the test duration was increased to 600 sec. Figure 8 is again referenced to show the bearing tester and bearing configuration. The increased test duration resulted in integrated levels of gamma ray dose and neutron flux of twice that experienced by the MIT-902 test.



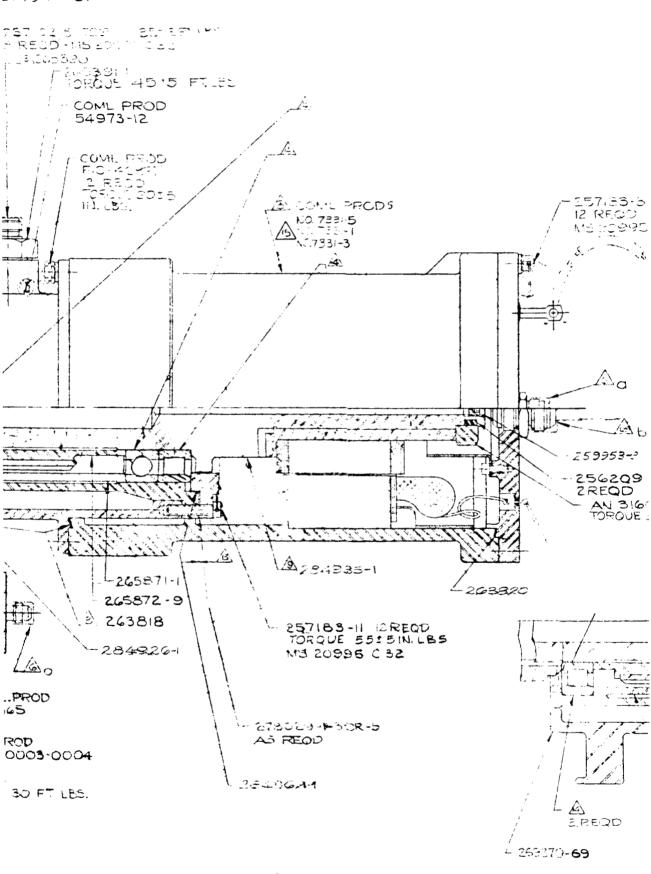


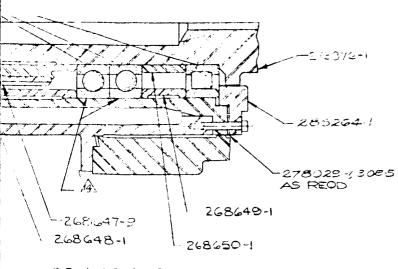
Figure 8

Refined Bearing Tester Assembly

FURCUE 9015IN LBS.

---2



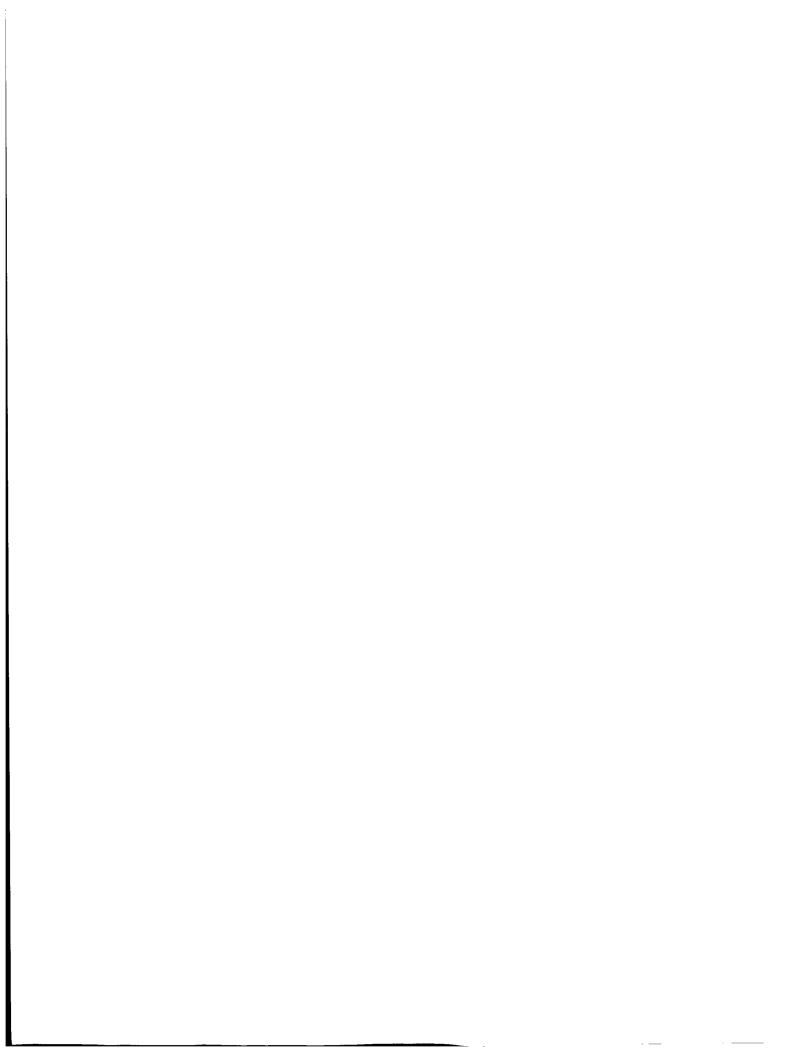


-29 4-39 ASSY

RN-S-0050

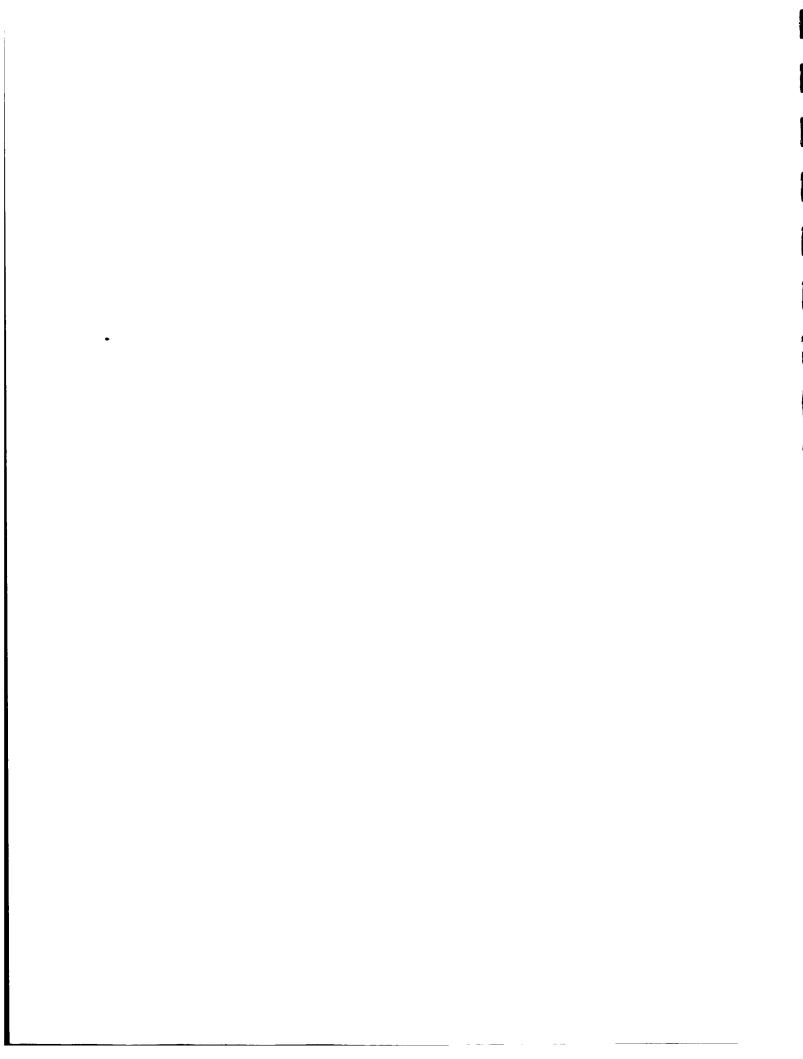
D. MIT-904

MIT-904 test hardware is, at present, unmodified from the MIT-902 and 903 configuration. The same pallet and bearing tester are to be used. However, a four, Armalon caged bearing configuration will be tested instead of the three bearing group tested in MIT-902 and 903. Figure 8 shows in the upper right hand corner the bearing configuration planned for the test. Test duration is planned to be 3600 sec, see Reference 9.



THEEEDING PAGE LINE NOT HEMEL.

IV RADIATION ANALYSIS



... LEULING PAGE LANGE NOT FILMED.

RN-S-0050

IV. RADIATION ANALYSIS

Basically, the techniques adopted for obtaining predicted radiation levels and heating rates were semi-empirical in nature. That is to say, the radiation sources were adjusted in their magnitude so that the calculated dose-rates at a given reference location were made to agree with measured data. This normalization to measurement was made at a distance of 12 inches from the test cell wall opposite the center of the core, for the case where the test cell was empty.

Having obtained sources for the calculable model which were designed to give the same result as a measurement in the unperturbed test cell, the analysis was performed by first mapping the remaining volume of the empty test cell and proceeding with alternate mockups of the test equipment. A recognized shortcoming of this technique is that all the sources are "lumped" into the active core. There are two areas of weakness in doing this. First, data calculated at points much closer to the core than the point of normalization are lower than measured values, since the capture gamma ray sources in the reactor structure (such as pressure vessel) have been juxtapositioned further from the detector location. This has been shown to represent no particular problem in conjunction with the tests under discussion, however, because points of interest are close to the point of normalization. The second weakness in this approach is that the effects of scattering from the test cell walls, and the existance of capture gamma ray sources in the walls, are not included. However, such contributions are difficult to analyze and quite often are not included in irradiation test evaluations. These contributions are made less important in the tests in question since the test equipment shields out some of these contributions as they affect such areas as the bearings. The scattered gamma radiation is particularly sensitive to this shielding effect since scattered gamma rays are so low in energy compared to direct radiation. leave out these indirect contributions has the effect of slightly underestimating the test environment in selected cases. This is much loss of a problem than overestimating the test environment in relating it to the NERVA flight condition.

TABLE 1

GAMMA RAY SOURCE
STRENGTH AND SPECTRUM

Group	Energy Internal (mev)	Effective Energy (mev)	Source Strength (mev/sec)
1	0 - 1.0	0.5	2.66 x 10 ¹⁸
2	1.0 - 2.0	1.5	1.59 x 10 ¹⁸
3	2.0 - 3.0	2.5	1.16 x 10 ¹⁸
14	3.0 - 5.0	4.0	5.02 x 10 ¹⁷
5	5.0 - 8.0	7.0	5.99 x 10 ¹⁷
The second point and the second secon		TOTAL	6.51 x 10 ¹⁸

A. GAMMA RADIATION ANALYSIS

The gamma ray analysis was based on a normalization of the calculated dose rate to the measured unpurturbed value at 12 inches from the test cell wall. The measured data was obtained from Table 3.2 of Reference (6). The calculated gamma ray spectrum was made to agree with the spectrum measured at a distance of 25 feet from the center of the core (as reported in Figure 3.4 and Table 3.9 of Reference (6). Figure 3.4 is reproduced here as Figure 9 for reference purposes. The result of this source normalization is indicated by the source strength spectrum presented in Table 1.

The QAD computer code (Reference 5) was used for gamma radiation calculations. Figure 10 shows how the calculated unpurturbed dose rates compare to the measured data of Reference (6). It will be noted in this Figure that the calculated values agree with the measured data from perfect agreement at the point of normalization to a condition where the calculated value is a factor of 2.65 below the measured value at a distance of 72 inches from the test cell wall. The reason for the divergence of the two curves is the inclusion of scattered and capture gamma ray contributions in the test cell walls in the measured results. However, as pointed out earlier, the self-shielding of the test specimens has the effect of markedly reducing the contribution of scattered gamma radiation. Scattered radiation is believed to be responsible for a very large portion of the discrepancy noted in the unperturbed traverse of Figure 10.

Unperturbed gamma dose rates were calculated at all points where perturbed test condition radiation intensities were desired. Each mechanical irradiation test configuration was then mocked up as a QAD computer model, and the test environment gamma dose rates and gamma heating rates were determined.

Pulse Height Spectrum of Gamma Rays from the 10 Mw ASTR $32\,$

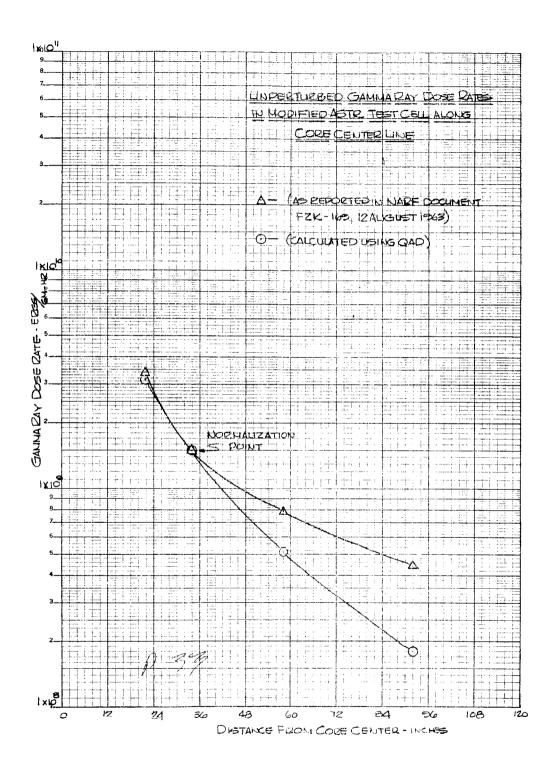


Figure 10
Unperturbed Gamma Ray Dose Rates in ASTR Test Cell

B. NEUTRON ANALYSIS

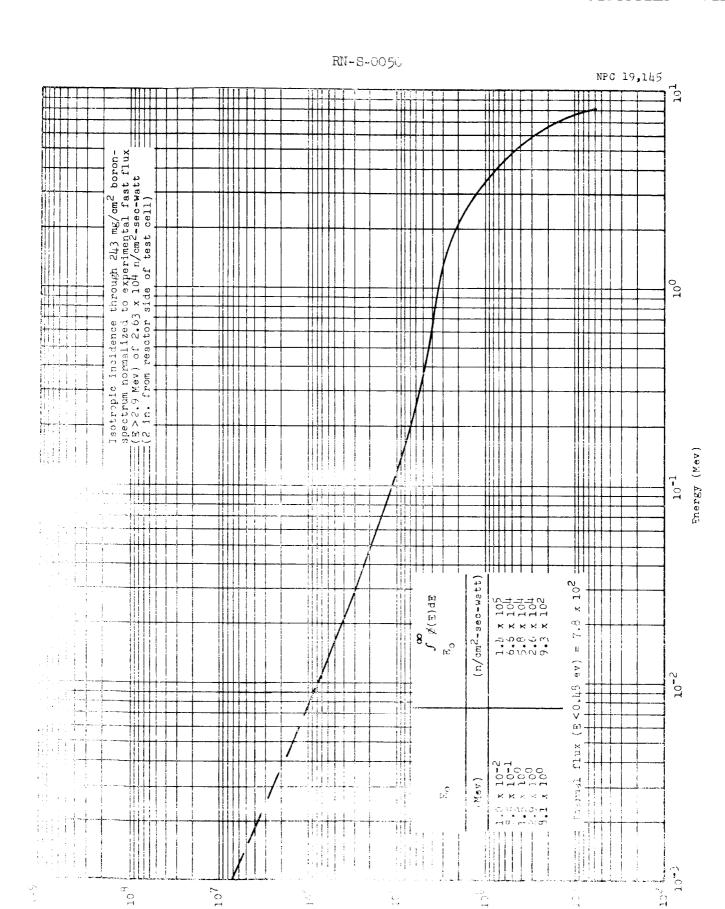
A similar approach was used in obtaining calculated neutron levels. First a normalization of source strength was completed such that the QAD calculated value (using effective removal theory) was made to agree with the measured neutron flux at a distance of 12 inches from the test cell wall. The neutron spectrum was based on the analytical spectrum reported in Figure 3.3 of Reference (6) and reported here as Figure 11.

An unperturbed neutron flux traverse was calculated along the test cell centerline, in the same manner as described in the gamma radiation analysis. Figure 12 shows this calculated traverse as it compares to the measured data of Reference (6). The fast neutron flux is based on a normalization to the measured flux above 0.85 Mw as reported in Table 3.2 of Reference (6). The fast neutron flux was then increased by a factor of 2.12 to include that portion of the spectrum shown in Figure 11 in the energy range from 0.01 Mw to 0.85 Mev. Final fast neutron flux values therefore are quoted above 0.01 Mev. These data then are directly comparable to NERVA data which is based on the same energy cut off point.

These calculations were based on a single energy group effective removal concept. The fast neutron source within the reactor was normalized to the measured 0.85 Mev threshold detector data and multiplied by a factor of 2.12 to include neutrons down to 0.01 Mev. The resultant source strength was 8.40 x 10^{17} n/sec for the 10 mw power level condition.

Unperturbed and perturbed (that is, test components in place) neutron fluxes above 0.01 Mev were calculated, using the QAD computer code, for all points of interest in the mechanical irradiation tests.

Figures 13, 14, and 15 show the analytical models used to describe the TSOV, TPCV, and bearing test for the QAD calculations.



April 13 mg 200 Car Sa Mgc College

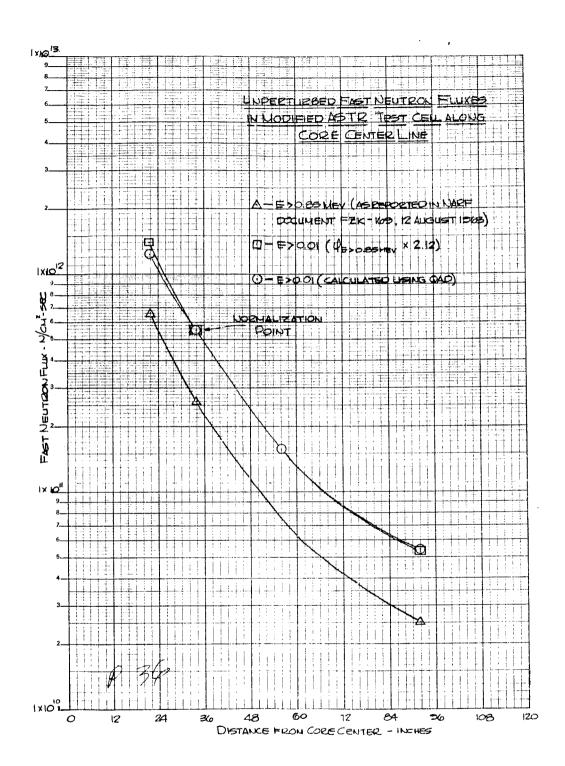
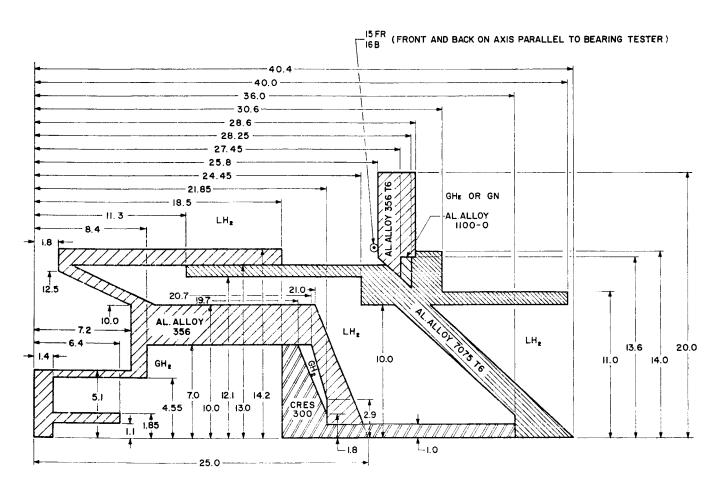


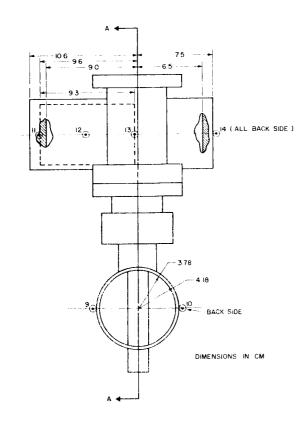
Figure 12
Unperturbed Fast Neutron Fluxes in ASTR Test Cell

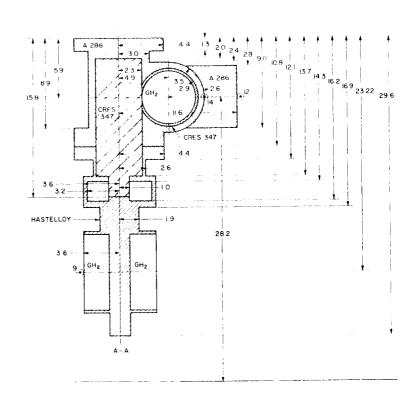


DIMENSION ARE IN CM.

Figure 13

Analytical Model - TSOV





TPCV
PART NUMBER 260563

Figure 14

Analytical Model - TPCV

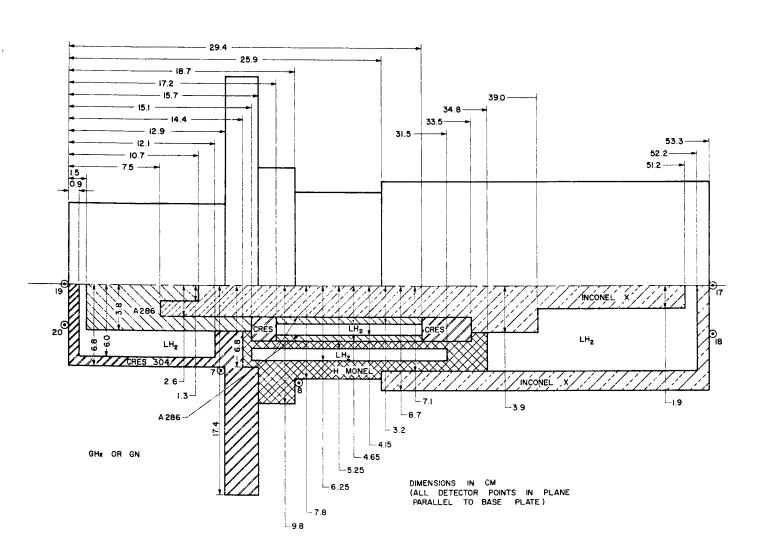
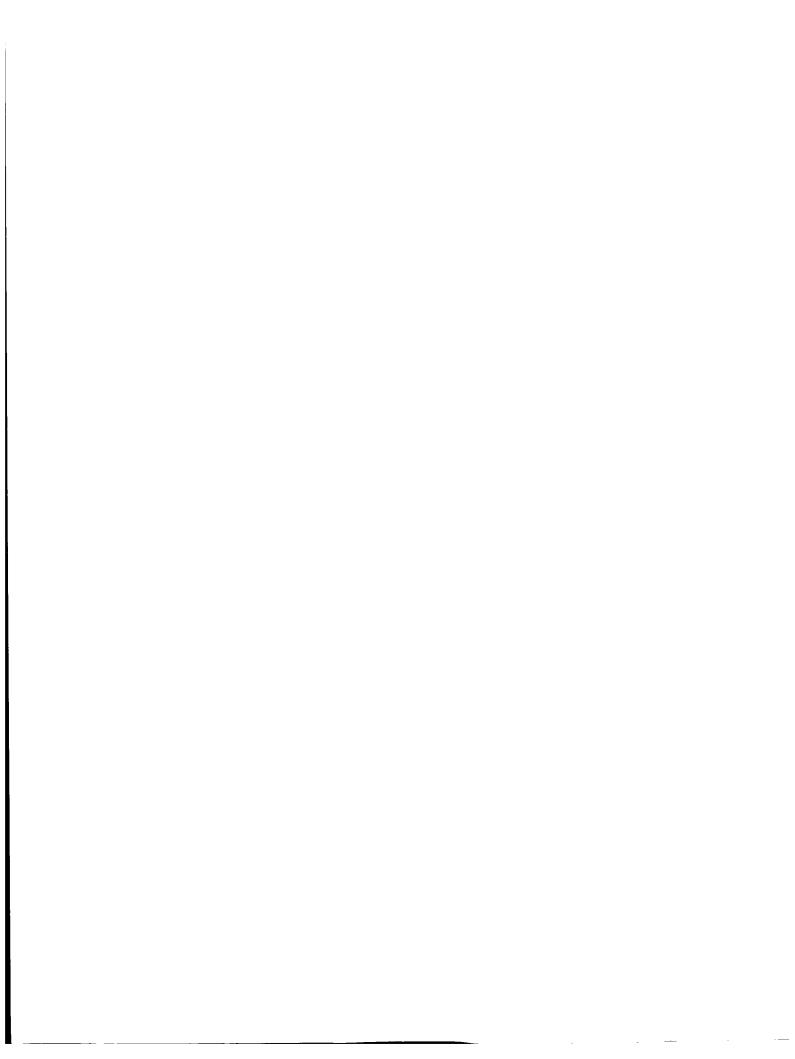
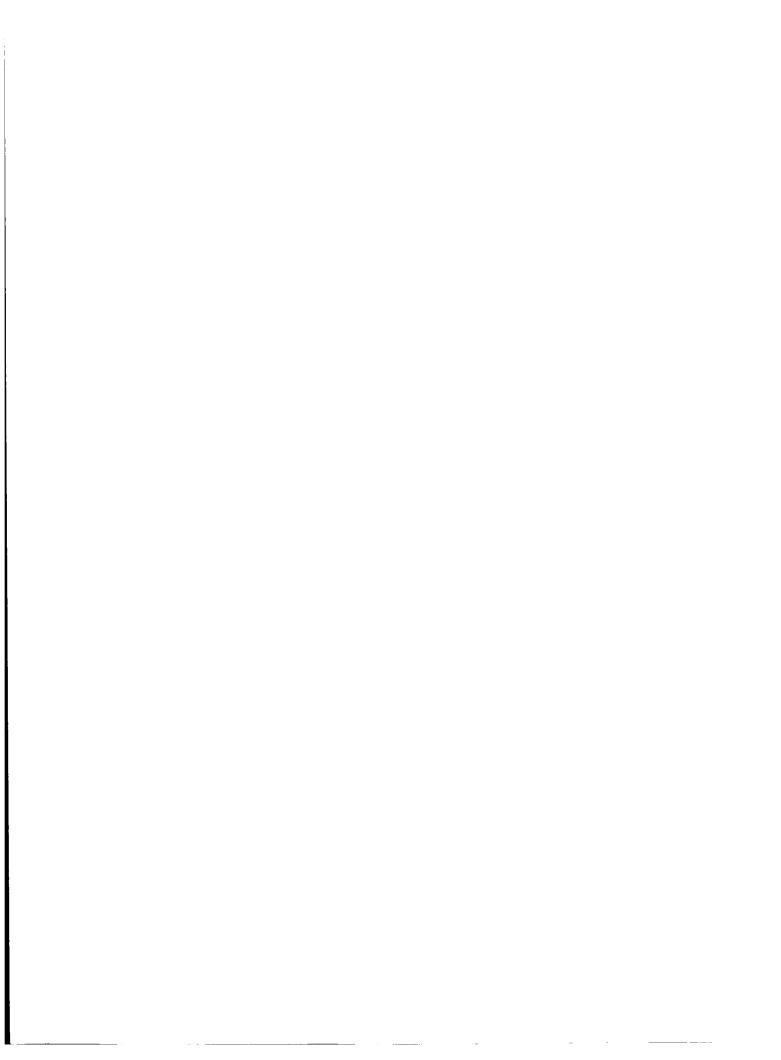


Figure 15
Analytical Model Bearing Tester



MECEDING PAGE DLANK NOT FILMED.

V CALCULATED RADIATION ENVIRONMENT



THECEDING PAGE DESIGN NOT FILMED.

RN-S-0050

V. CALCULATED RADIATION ENVIRONMENT

The results of the QAD computer program calculations are presented in this section. Neutron and gamma radiation intensities are described along with gamma heating rates. In addition, the neutron-to-gamma number flux ratio is also reported for a number of dose-point locations in the various MIT tests. Finally, comparisons are made with comparable NERVA flight environment conditions as they are known at the time of this writing.

A. MIT-901

Table 2 presents the results of the analysis of the MIT-901 radiation environment. Gamma dose rates are reported in units of (ergs/gm) of carbon per hour at the ASTR 10 Mw power level. Integrated gamma doses are given in terms of (ergs/gm) of carbon for a total exposure time of 327 seconds. Gamma ray number fluxes are given in units of $(\gamma/\text{cm}^2\text{sec})$. Fast neutron fluxes for E >0.01 MeV are reported in the following manner: two values are given: an unperturbed flux with no test configuration in place, and a perturbed flux as calculated with effective removal theory. Removal theory is known to over-estimate the neutron attenuation of metals in a condition such as the present one. Therefore, the most probable flux level is somewhere in the range given for each component. The unperturbed flux will hereafter be referred to as the maximum value in each case. Two n/γ flux ratios are given in correspondence with the two neutron flux values. The heating rates are in terms of (BTU/sec) per cubic inch of the material in which the detector is located. Table 3 provides a comparison between the data for the ASTR test and the current predictions of the NERVA flight conditions. This comparison is limited to the TSOV, TPCV, and turbopump bearings because they represent the only components which have spatically defined locations in NERVA.

Figure 16 depicts the location of the dose points in the MIT-901 analysis.

This test has been completed. Comparisons with some test results are given in a later section.

RADIATION ENVIRONMENT AND HEATING RATES FOR MECHANICAL TEST - 901 (327 sec Duration)

			TEST - 901 (3c) see Duraction)					
Detector Point	Detector Location	Material at Detector Location	Gamma Ray Dose-Rate (ergs/gm(c)hr)	Int. Gamma Ray Dose (ergs/gm(c))	Gamma Ray No. Flux (7/cm²/sec)	Fast Neutron Flux E 0.01 Mev (n/cm2-sec)	Neutron/Gamma No. Flux Ratio	Gamma Heating Rate (BTU/in ³ -sec)
1	Flange and seal test - Al to Al flange	Al 6061	7.8 x 10 ⁸	7.1 x 10 ⁷	3.8 x 10 ¹²	1.1 - 1.7 x 10 ¹¹ 2.4 - 3.5 x 10 ¹¹	0.029 - 0.044	8.9 x 10 ⁻⁴
5	Flange and seal test - Al to SS flange	Al 6061	1.2 x 10 ⁹	1.1 x 10 ⁸	5.8 x 10 ¹²		0.041 - 0.060	1.3 x 10 ⁻³
3	Flange and seal test - SS to SS test	SS 347	1.8 × 10 ⁹	1.6 x 10 ⁸	8.7 x 10 ¹²	4.7 - 5.2 x 10 ¹¹	0.054 - 0.060	6.2 x 10 ⁻³
4	Bearing test	SS 302	6.4 x 10 ⁸	5.8 x 10 ⁷	3.1 x 10 ¹²	2.4 - 6.0 x 10 ¹¹	0.077 - 0.19	2.2 x 10 ⁻³
5	Bearing test	Inconel X	4.7 x 10 ⁷	4.3 x 10 ⁶	2.3 x 10 ¹¹	0.6 - 5.7 x 10 ¹¹	0.26 - 2.5	2.0 x 10 ⁻⁴
6	Bearing test	SS 302	3.5 x 10 ⁸	3.2 x 10 ⁷	1.7 x 10 ¹²	1.7 - 5.5 x 10 ¹¹	0.10 - 0.32	1.3 x 10 ⁻³
7	Bearing test - Pump end bearing	SS 302	1.8 x 10 ⁸	1.6 x 10 ⁷	6.7 x 10 ¹¹	1.0 - 5.6 x 10 ¹¹	0.11 - 0.64	7.5 x 10 ⁻¹⁴
8	Bearing test	Inconel X	3.1 x 10 ⁷	2.8 x 10 ⁶	1.5 x 10 ¹¹	0.4 - 5.2 x 10 ¹¹	0.27 - 3.5	1.2 x 10 ⁻⁴
9	Bearing test	H Monel	1.8 x 10 ⁸	1.6 x 10 ⁷	8.7 x 10 ¹¹	1.3 - 5.0 x 10 ¹¹	0.15 - 0.57	6.5 x 10 ⁻¹⁴
10	Bearing test	Inconel X	8.5 x 10 ⁶	7.7 x 10 ⁵	4.1 x 10 ¹⁰	0.2 - 4.2 x 10 ¹¹	0.49 - 10	3.4 x 10 ⁻⁵
11	Bearing test	A 286	6.8 x 10 ⁷	6.2 x 10 ⁶	3.3 x 10 ¹¹	0.4 - 4.2 x 10 ¹¹	0.12 - 1.3	2.8 x 10 ⁻⁴
12	Bearing test	Inconel X	7.7 x 10 ⁷	7.0 x 10 ⁶	3.7 x 10 ¹¹	0.6 - 4.2 x 10 ¹¹	0.16 - 1.1	2.8 x 10 ⁻⁴
13	Bearing test - Turbine end bearing	SS 302	2.3 x 10 ⁷	2.1 x 10 ⁶	1.1 x 10 ¹¹	0.2 - 3.9 x 10 ¹¹	0.18 - 3.5	9.8 x 10 ⁻⁵
14	Bearing test - Turbine end bearing	SS 302	2.3 x 10 ⁷	2.1 x 10 ⁶	1.1 x 10 ¹¹	$0.2 - 3.7 \times 10^{11}$	0.18 - 3.4	9.9 x 10 ⁻⁵
15	Bearing test	Inconel X	3.5 x 10 ⁶	3.2 x 10 ⁵	1.7 x 10 ¹⁰	0.08 - 3.5 x 10 ¹¹	0.47 - 22	1.4 x 10 ⁻⁵
16	Bearing test - Turbine end bearing	SS 302	8.1 x 10 ⁶	7.4 x 10 ⁵	3.9 x 10 ¹⁰	0.1 - 3.6 x 10 ¹¹	0.26 - 9.2	3.6 x 10 ⁻⁵
17	Bearing test	H Monel	1.4 x 10 ⁷	1.3 x 10 ⁶	6.8 x 10 ¹⁰	0.2 - 3.5 x 10 ¹¹	0.29 - 5.1	5.5 x 10 ⁻⁵
18	Bearing test	Inconel X	8.4 x 10 ⁵	7.6 x 10 ⁴	4.1 x 10 ⁹	0.02 - 2.3 x 10 ¹¹	0.49 - 56	3.4 x 10 ⁻⁶
19	Bearing test	Inconel X	5.2 x 10 ⁶	4.7 x 10 ⁵	2.5 x 10 ¹⁰	0.06 - 2.4 x 10 ¹¹	0.24 - 9.6	2.0 x 10 ⁻⁵
20	TPCV	Hastelloy B	1.1 x 10 ⁹	1.0 x 10 ⁸	5.3 x 10 ¹²	3.0 - 3.5 x 10 ¹¹	0.057 - 0.066	4.2 x 10 ⁻³
21	TSOV	A 356	4.9 x 10.8	4.4 x 10 ⁷	2.4 x 10 ¹²	1.2 - 2.2 x 10 ¹¹	0.050 - 0.092	5.6 x 10 ⁻¹⁴

Table 2
Radiation Environment - MIT 901

CONFIDENTIAL — RESTRICTED DATA

RN-S-0050

TABLE 5

COMPARISON OF NERVA AND 901 TEST ENVIRONMENTS

A. MERVA - (1300 SMC)

Component	Maximum y - Dusc Rate (erg/gm-hr)	Maximum γ - Dose (erg/gm)	Maximum γ - Fiux (γ/cm-sec)	Maximum n-Flux E > J _o oi Mev (n/cm ² sec)	Meximum nyt (n/cm²)	Maximum n/γ Flux Ratio	Maximum Gamma Heating Rate (BTU/in-sec)
TSOV	6.0 x 10 ⁷	2.0 x 10 ⁷	1.8 x 10 ¹¹	3.0 x 10 ¹⁰	3.6 x 10 ¹³	0.17	2.0 x 10 ⁻⁵
TPCV	1.4 x 10 ⁹	4.7 x 10 ⁸	4.6 x 10 ¹²	4.0 x 10 ¹¹	4.8 x 10 ¹⁴	0.09	5.3 x 10 ⁻³
BEARINGS, PUMP END	1.1 x 10 ⁷	3.7 x 10 ⁶	10 3.4 x 10	1.0 x 10 ¹¹	1.2 x 10 ¹⁴	2.9	5.9 x 10 ⁻⁵
BEARINGS, TURBINE END	1.4 x 10 ⁸	4.7 × 10 ⁷	4.3 x 10 ¹¹	1.0 x 10 ¹¹	1.2 x 10 ¹⁴	0.23	7.5 x 10 ⁻⁴

Component	Maximum γ - Dose Rate (erg/gm-hr)	Maximum y - Dose (erg/gm)	Maximum 7 - Flux (y/cm ² sec)	Muximum n-Flux E > 0.91 Mev (n/cm-sec)	Maximum nvt (n/cm ²)	Maximum n/γ Flux Ratio	Maximum Gamma Heating Rate (BTU/in ² sec)
TSOV	4.9 x 10 ⁸	4.4 x 10 ⁷	2.4 x 10 ¹²	2.2 x 10 ¹¹	7.2 x 10 ¹³	0.098	5.6 x 10 ⁻⁴
TPCV	1.1 x 10 ⁹	1.0 x 10 ⁸	5.3 x 10 ¹²	5.5 x 10 ¹¹	1.1 x 10 ¹⁴	0.066	4.2 x 10 ⁻³
BEARINGS, PUMP END	1.8 x 10 ⁸	1.6×10^{7}	8.7 x 10 ¹¹	5.6 x 10 ¹¹	1.8 x 10 ¹⁴	0.64	7.5 x 10 ⁻ⁱ⁴
BEARINGS, TURBINE END	2.3 x 10 ⁷	2.1x 10 ⁵	1.1×10^{10}	5.9 x 10 ¹¹	1.3 x 10 ¹⁴	3.4	9.9 x 10 ⁻⁵

Table 3

Comparison of NERVA and ASTR - 901 Test Environments

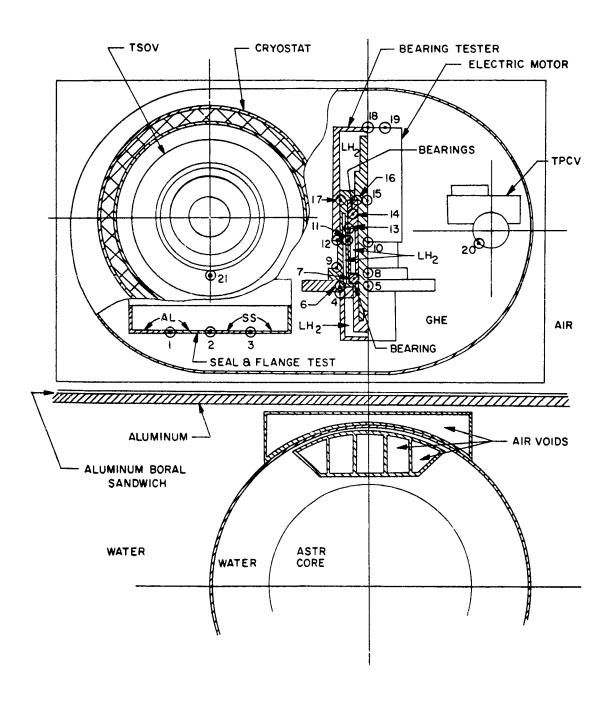


Figure 16

Dose Point Location for MIT 901

B. MIT-902

Table 4 is essentially repeated for the bearing tester configuration tested in MIT-902. The dose rates and neutron and gamma fluxes are not very different from similar locations in the 901 test. The run duration in this case was 300 seconds.

 $\hbox{ Comparisons are made between these calculated data and NERVA in Table 5. } \\$

This test has been completed and some comparisons are available with measured data, as given in Section VI of this report.

A. RADIATION ENVIRONMENT AND HEATING RATES FOR MECHANICAL IRRADIATION TEST 902 (300 SEC DURATION)

Front rate of Stream	DETECTOR POINT		MATERIAL AT DETECTOR LOCATION	GAMMA RAY DOSE RATE (ergs/gm(c)-hr)	INTEGRATED GAMMA FAY DOSE (ergs/gm(c)	GAMMA RAY NUMBER FLUX (7 /cm²-sec)	FAST NEUTRON FLUX E > 0,01 Mev (n/cm²-sec)	NEUTRON/GAMMA NUMBER FLUX RATIO	MAXIMUM ₂ NVT-n/cm	GAMMA RAY HEATING RATE BTU/in ³ -sec
Front of bearing section Size 2.6 x 10^7 2.3 x 10^2 1.4 x 10^1 5.8 x 9.8 x 10^1 0.043 - 0.07 2.9 x 10^1 5.8 t 6.8 x 10 1.5 x 10^1 5.8 x 10 1.5 x 10^1 5.8 x 10 1.5 x 10^1 5.8 x 10^1	601		Al 6061	3.6 x 10 ⁹	3.0 × 10 ⁸	1.7 x 10 ¹³	1.2 - 1.2 x 10 ¹²	0.07 - 0.07	3.6 x 10 ¹⁴	4.0 x 10 ⁻³
Top, center, forward end of bearing section Nome 1.7 x 10 ⁸ 1.4 x 10 ⁷ 8.2 x 10 ¹¹ 1.0 - 4.9 x 10 ¹¹ 0.12 - 6.60 1.5 x 10 ³¹ 6.8 x 10 ⁷ Pupp end bearing 83 302 2.8 x 10 ⁷ 2.3 x 10 ⁵ 1.4 x 10 ¹¹ 0.7 - 5.6 x 10 ¹¹ 0.20 - 6.0 1.7 x 10 ¹¹ 1.1 x 10	614		ss 302	2.8 x 10 ⁹	2.3 × 10 ⁸	1.4 x 10 ¹³	8.8 - 9.8 x 10 ¹¹	0.063 - 0.07	2.9 x 10 ¹⁴	9.7 x 10 ⁻³
13 Turbine and bearing 85 302 1.0 x 10 ⁷ 8.3 x 10 ⁵ 1.9 x 10 ¹⁰ 0.007 3.9 x 10 ¹¹ 0.20 8.1 1.2 x 10 ¹¹ 1.0 x 10 ¹ 1.0 x	615		H Monel	1.7 x 10 ⁸	1.4 x 10 ⁷	8.2 x 10 ¹¹	1.0 - 4.9 x 10 ¹¹	0.12 - 0.60	1.5 × 10 ¹⁴	6.8 x 10 ⁻⁴
14 Turbine end bearing	7	Pump end bearing	SS 302	2.8 x 10 ⁷	2.3 × 10 ⁶	1.4 x 10 ¹¹	0.37 - 5.6 x 10 ¹¹	0.26 - 4.0	1.7 x 10 ¹⁴	1.1 x 10 ⁻⁴
16 Marking on Descring \$3 302 1.7 x 10 ⁵ 1.8 x 10 ⁵ 9.2 x 10 ³ 0.90 - 3.6 x 10 ¹¹ 0.49 - 10 1.1 x 10 ¹⁴ 6.9 x 10 ¹³ 5.6 x 10 ¹⁴ 0.49 - 10 0.4	13	Turbine end bearing	SS 302	1.0 × 10 ⁷	8.3 x 10 ⁵	4.8 x 10 ¹⁰	0.097 - 3.9 x 10 ¹¹	0.20 - 8.1	1.2 x 10 ¹⁴	4.0 x 10 ⁻⁵
Reactor centerline	1,4	Turbine end bearing	SS 302	4.0 x 10 ⁶	3.3 × 10 ⁵	1.9 x 10 ¹⁰		0.35 - 19	1.1 × 10 ¹⁴	1.6 x 10 ⁻⁵
610 Reactor centerline- Form of boaring tester Income X 1.1 x 10 ⁵ 9.2 x 10 ¹ 5.3 x 10 ⁹ 0.0062 - 2.3 x 10 ¹¹ 0.79 - 130 5.2 x 10 ¹³ 3.7 x 10 ¹³ 3.1 x 10 ⁵ 2.6 x 10 ⁴ 1.5 x 10 ⁹ 0.0063 - 1.9 x 10 ¹¹ 0.36 - 130 5.7 x 10 ¹³ 3.7 x 10 ¹³ 3.7 x 10 ¹³ 3.1 x 10 ⁵ 3.1 x 10 ⁵ 6.0 x 10 ⁶ 1.7 x 10 ¹³ 1.2 - 1.2 x 10 ¹² 0.07 - 0.07 7.2 x 10 ¹⁴ 8.0 x 10 ⁶ 8 mactor centerline- form of face of whrous 5.0 c 2.8 x 10 ⁹ 1.6 x 10 ⁶ 1.4 x 10 ¹³ 8.6 · 9.8 x 10 ¹¹ 0.063 - 0.07 5.6 x 10 ¹⁴ 1.9 x 10 ¹⁵ 7.0 c 4.9 x 10 ¹¹ 0.12 - 0.60 2.9 x 10 ¹⁴ 1.4 x 10 ¹⁵ 7.0 c 4.9 x 10 ¹¹ 0.12 - 0.60 2.9 x 10 ¹⁴ 1.4 x 10 ¹⁵ 7.0 c 4.8 x 10 ¹⁵ 7.0 c 4.8 x 10 ¹⁵ 0.067 - 3.9 x 10 ¹¹ 0.65 - 1.0 3.4 x 10 ¹⁴ 0.2 x 10 ¹⁵ 1.5 x 10 ¹⁵ 1.5 x 10 ¹⁵ 1.5 x 10 ¹⁵ 1.7 x 10 ¹⁵ 1.5 x 10 ¹⁵ 1.7 x 10 ¹⁵ 1.2 x 10 ¹⁵ 1.2 x 10 ¹⁵ 0.00 - 3.6 x 10 ¹¹ 0.2 x 10 ¹⁵ 1.4 x 10 ¹⁵ 1.5 x 10 ¹⁵ 1.5 x 10 ¹⁵ 1.7 x 10 ¹⁵	16	Turbine end bearing	SS 302	1.7 x 10 ⁶	1.4 x 10 ⁵	8.2 x 10 ⁹	0.040 - 3.6 x 10 ¹¹	0.49 - 44	1.1 × 10 ¹⁴	6.9 x 10 ⁻⁶
Reactor centerline-	616 (18)		Inconel X	1.1 × 10 ⁵	9.2 x 10 ³	5.3 x 10 ⁸	0.0042 - 2.3 x 10 ¹¹	0.79 - 430	6.9 x 10 ¹³	5.2 x 10 ⁻⁷
### RECENTICAL IMPRAISATION TEST 903 (600 SEC UMMATION) Searctor centerlines	610		Al 6061	3.1 x 10 ⁵	2,6 x 10 ⁴	1.5 x 10 ⁹	0.0068 - 1.9 x 10 ¹¹	0.36 - 130	5.7 x 10 ¹³	3.7 x 10 ⁻⁷
601 Reactor centerline- Front face of shroud 614 Reactor centerline- Front face of shroud 615 Stock face of shroud 616 Reactor centerline- Front face of shroud 617 Reactor centerline- Front face of shroud 618 Reactor centerline- Front face of shroud 619 Reactor centerline- Front face of shroud 619 Reactor centerline- Front face of shroud 610 Reactor centerline- Front face of shroud 610 Reactor centerline- Front face of shroud 610 Reactor centerline- Front face of shroud 611 Reactor centerline- Front face of shroud 612 Reactor centerline- Front face of shroud 613 Reactor centerline- Front face of shroud 614 Reactor centerline- Front face of shroud 615 Reactor centerline- Front face of shroud 616 Reactor centerline- Front face of shroud 617 Reactor centerline- Front face of shroud 618 Reactor centerline- Front face of shroud 619 Reactor centerline- Front face of shroud 610 Reactor centerline- Front face of shroud 610 Reactor centerline- Front face of shroud 610 Reactor centerline- Front face of shroud 611 Reactor centerline- Front face of shroud 612 Reactor centerline- Front face of shroud 613 Reactor centerline- Front face of shroud 614 Reactor centerline- Front face of shroud 615 Reactor centerline- Front face of shroud 616 Reactor centerline- Front face of shroud 617 Reactor centerline- Front face of shroud 618 Reactor centerline- Front face of shroud 619 Reactor centerline- Front face of shroud 610 Reactor centerline- Front face of shroud 610 Reactor centerline- Front face of shroud 610 Reactor centerline- Front face of shroud 611 Reactor centerline- Front face of shroud 612 Reactor centerline- Front face of shroud 613 Reactor centerline- Front face of shroud 614 Reactor centerline- Front face of shroud 615 Reactor centerline- Front face of shroud 616 Reactor centerline- Front face of shroud 617 Reactor centerline- Front face of shroud 618 Reactor centerline- Front face of shroud 619 Reactor centerline- Front face of shroud 619 Reactor centerline- Front face of shroud 610 Reactor centerline- Front face of shr							(NO			
## Comparing tester Comparing tester Comparing tester Comparing tester Comparing section Compari	601		A1 6061	_		•		0.07 - 0.07	7.2 x 10 ¹⁴	8.0 x 10 ⁻³
Pair of bearing action 7 Pump end bearing SS 302 2.8 x 10 ⁷ 1.7 x 10 ⁶ 1.4 x 10 ¹¹ 0.37 - 5.6 x 10 ¹¹ 0.26 - 4.0 3.4 x 10 ¹¹ 2.2 x 10 ¹¹ 13 Turbine end bearing SS 302 1.0 x 10 ⁷ 1.7 x 10 ⁶ 4.8 x 10 ¹⁰ 0.097 - 3.9 x 10 ¹¹ 0.20 - 6.1 2.3 x 10 ¹¹ 8.0 x 10 ¹¹ 14 Turbine end bearing SS 302 4.0 x 10 ⁶ 6.6 x 10 ⁵ 1.9 x 10 ¹⁰ 0.067 - 3.7 x 10 ¹¹ 0.35 - 19 2.2 x 10 ¹¹ 1.4 x 10 ¹¹ 16 Turbine end bearing SS 302 1.7 x 10 ⁶ 2.8 x 10 ⁵ 8.2 x 10 ⁹ 0.040 - 3.6 x 10 ¹¹ 0.49 - 44 2.2 x 10 ¹¹ 1.4 x 10 ¹¹ 16 Turbine end bearing Entering Income X 1.1 x 10 ⁵ 1.8 x 10 ⁶ 5.3 x 10 ⁸ 0.0042 - 2.3 x 10 ¹¹ 0.79 - 430 1.4 x 10 ¹¹ 1.0 x 10 ¹¹ 17 Reactor centerline-rear face of shroud 18 Reactor centerline-front face of shroud 19 Reactor centerline-front face of shroud 20 Reactor centerline-front face of shroud 21 Reactor centerline-front face of shroud 22 Reactor centerline-front face of shroud 31 x 10 ⁹ 2.8 x 10 ⁹ 1.7 x 10 ¹³ 1.2 - 1.2 x 10 ¹² 0.07 - 0.07 4.3 x 10 ¹⁵ 1.2 x	614			2.8 x 10 ⁹	4.6 x 10 ⁸	1.4 x 10 ¹³	8.8 - 9.8 x 10 ¹¹	0.063 - 0.07	5.8 x 10 ¹⁴	1.9 x 10 ⁻²
13 Turbine end bearing SS 302 1.0 x 10 ⁷ 1.7 x 10 ⁶ 4.8 x 10 ¹⁰ 0.097 - 3.9 x 10 ¹¹ 0.20 - 8.1 2.3 x 10 ¹¹ 8.0 x 10 ¹¹ 14 Turbine end bearing SS 302 4.0 x 10 ⁶ 6.6 x 10 ⁵ 1.9 x 10 ¹⁰ 0.067 - 3.7 x 10 ¹¹ 0.35 - 19 2.2 x 10 ¹¹ 3.2 x 10 ¹¹ 10.0 x 10 ¹¹ 1.4 x 10 ¹¹ 1.6 x 10 ¹¹ 1.4 x 10 ¹¹ 1.4 x 10 ¹¹ 1.4 x 10 ¹¹ 1.6 x 10 ¹¹ 1.5 x 10 ¹¹ 1.5 x 10 ¹¹ 0.004 - 3.6 x 10 ¹¹ 0.09 - 44 2.2 x 10 ¹¹ 1.4 x 10 ¹¹ 1.6 x 10 ¹¹ 1.6 x 10 ¹¹ 1.5 x 10 ¹¹ 0.0068 - 1.9 x 10 ¹¹ 0.36 - 130 1.4 x 10 ¹¹ 1.4 x 10 ¹¹ 1.5 x 10 ¹¹ 1.	615			1.7 × 10 ⁸	2.8 x 10 ⁷	8.2 x 10 ¹¹	1.0 - 4.9 x 10 ¹¹	0.12 - 0.60	2.9 x 10 ¹⁴	1.4 x 10 ⁻³
14 Turbine end bearing SS 302 4.0 x 10 ⁶ 6.6 x 10 ⁵ 1.9 x 10 ¹⁰ 0.067 - 3.7 x 10 ¹¹ 0.35 - 19 2.2 x 10 ¹⁴ 3.2 x 10 ¹⁶ 1.6 x 10 ⁶ 2.8 x 10 ⁵ 8.2 x 10 ⁹ 0.040 - 3.6 x 10 ¹¹ 0.49 - 44 2.2 x 10 ¹⁴ 1.4 x 10 ¹⁶ 1.6 (18) Reactor centerline-rear face of shroud 610 Reactor centerline-rear face of shroud 610 Reactor centerline-rear face of shroud 611 Reactor centerline-rear face of shroud 612 Reactor centerline-rear face of shroud 613 Reactor centerline-rear face of shroud 614 Reactor centerline-front face of shroud 615 Top, center, forward end of bearing tester 616 Top, center, forward end of bearing section 7 Pump end bearing 617 SS 302 2.8 x 10 ⁷ 2.8 x 10 ⁷ 1.4 x 10 ¹¹ 1.0 - 4.9 x 10 ¹¹ 0.12 - 0.60 1.8 x 10 ¹⁵ 8.1 x 10 ¹⁵ 1.3 x 10 ¹⁵ 1.0 x 10 ⁷ 1.0 x 10 ⁷ 1.0 x 10 ⁷ 1.4 x 10 ¹¹ 0.37 - 5.6 x 10 ¹¹ 0.20 - 8.1 1.4 x 10 ¹⁵ 1.3	7	Pump end bearing	SS 302	2.8×10^{7}	4.6 x 10 ⁶	1.4 x 10 ¹¹	$0.37 - 5.6 \times 10^{11}$	0.26 - 4.0	3.4 x 10 ¹⁴	2.2 x 10 ⁻⁴
16 Turbine end bearing SS 302 1.7 x 10 ⁶ 2.8 x 10 ⁵ 8.2 x 10 ⁹ 0.040 - 3.6 x 10 ¹¹ 0.49 - 44 2.2 x 10 ¹⁴ 1.4 x 10 ¹⁴ 616 (18) Reactor centerline-rear face of shroud Reactor centerline-rear face of shroud C. RADIATION ENVIRONMENT AND REATING RATES FOR MECHANICAL IMPADIATION IEST 90 ⁴ (3600 SEC DURATION) C. RADIATION ENVIRONMENT AND REATING RATES FOR MECHANICAL IMPADIATION IEST 90 ⁴ (3600 SEC DURATION) C. RADIATION ENVIRONMENT AND REATING RATES FOR MECHANICAL IMPADIATION IEST 90 ⁴ (3600 SEC DURATION) C. RADIATION ENVIRONMENT AND REATING RATES FOR MECHANICAL IMPADIATION IEST 90 ⁴ (3600 SEC DURATION) C. RADIATION ENVIRONMENT AND REATING RATES FOR MECHANICAL IMPADIATION IEST 90 ⁴ (3600 SEC DURATION) C. RADIATION ENVIRONMENT AND REATING RATES FOR MECHANICAL IMPADIATION IEST 90 ⁴ (3600 SEC DURATION) C. REACTOR CENTERLINE- SS 30 ² 2.8 x 10 ⁹ 3.6 x 10 ⁹ 1.7 x 10 ¹³ 1.2 - 1.2 x 10 ¹² 0.07 - 0.07 4.3 x 10 ¹⁵ 4.8 s 10 ¹⁵ 1.2	13	Turbine end bearing	SS 302		1.7×10^6			0.20 - 8.1	2.3 x 10 ¹⁴	8.0 x 10 ⁻⁵
616 (18) Reactor centerline-rear face of shroud Reactor centerline-rear face of shrou	14	Turbine end bearing	SS 302	4.0 x 10 ⁶	6.6 x 10 ⁵	1.9 x 10 ¹⁰	0.067 - 3.7 x 10 ¹¹	0.35 - 19	2.2 x 10 ¹⁴	3.2 x 10 ⁻⁵
Reactor centerline-rear face of shroud C. RADIATION ENVIRONMENT AND REATING RATES FOR MECHANICAL TRRADIATION IEST 904 (3600 SEC IMPATION) 601 Reactor centerline-front face of shroud 614 Reactor centerline-front of bearing tester 615 Top, center, forward end of bearing section 7 Pump end bearing SS 302 2.8 x 10 ⁷ 2.8 x 10 ⁷ 2.8 x 10 ⁷ 3.6 x 10 ⁷ 3.6 x 10 ⁷ 3.6 x 10 ⁸ 3.7 x 10 ⁸ 3.7 x 10 ⁸ 3.7 x 10 ⁸ 3.8 x 10 ¹¹ 3.0 x 10 ¹¹ 3.1 x 10 ¹² 3.6 x 10 ⁹ 3.6 x 10 ¹³ 3.7 x 10 ¹³ 3.8 x 10 ¹² 3.9 x 10 ¹¹ 3.1 x 10 ¹² 3.5 x 10 ¹⁵ 3.5 x 10 ¹⁵ 3.6 x 10 ⁹ 3.6 x 10 ¹³ 3.7 x 10 ¹³ 3.8 x 10 ¹¹ 3.9 x 10 ¹¹ 3.1 x 10 ¹⁵ 3.1 x	16	Turbine end bearing	SS 302	1.7 x 10 ⁶	2.8 x 10 ⁵		0.040 - 3.6 x 10 ¹¹	0.49 - 44	2.2 x 10 ¹⁴	1.4 x 10 ⁻⁵
C. RADIATION ENVIRONMENT AND REATING RATES FOR MECHANICAL IRRADIATION TEST 904 (3600 SEC BURATION) 601 Reactor centerline-front face of shroud 614 Reactor centerline-front of bearing sestion 615 Top, center, forward end of bearing section 7 Pump end bearing 85 302 2.8 x 10 ⁷ 2.8 x 10 ⁷ 1.4 x 10 ¹¹ 0.37 - 5.6 x 10 ¹¹ 0.26 - 4.0 1.7 x 10 ⁸ 1.7 x 10 ⁸ 8.2 x 10 ¹¹ 0.37 - 5.6 x 10 ¹¹ 0.20 - 8.1 1.4 Turbine end bearing 85 302 1.0 x 10 ⁷ 1.0 x 10 ⁷ 4.8 x 10 ¹⁰ 0.067 - 3.7 x 10 ¹¹ 0.35 - 19 1.3 Turbine end bearing 85 302 4.0 x 10 ⁶ 4.0 x 10 ⁶ 1.9 x 10 ¹⁰ 0.067 - 3.7 x 10 ¹¹ 0.35 - 19 1.3 x 10 ¹⁵ 1.3 x 10 ¹⁵ 1.9 x 10 ¹⁶ 1.7 x 10 ⁶ 8.2 x 10 ⁹ 0.040 - 3.6 x 10 ¹¹ 0.49 - 44 1.4 Turbine end bearing 85 302 1.7 x 10 ⁶ 1.7 x 10 ⁶ 8.2 x 10 ⁹ 0.040 - 3.6 x 10 ¹¹ 0.49 - 44 1.3 x 10 ¹⁵ 8.3 x 10 ¹⁵ 1.9 x 10 ¹⁶ 1.7 x 10 ⁶ 8.2 x 10 ⁹ 0.040 - 3.6 x 10 ¹¹ 0.49 - 44 1.3 x 10 ¹⁵ 8.3 x 10 ¹⁵ 1.9 x 10 ¹⁶ 1.7 x 10 ⁶ 8.2 x 10 ⁹ 0.040 - 3.6 x 10 ¹¹ 0.49 - 44 1.3 x 10 ¹⁵ 8.3 x 10 ¹⁶ 1.9 x 10 ¹⁶ 1.1 x 10 ⁵ 1.1 x 10 ⁵ 1.1 x 10 ⁵ 1.5 x 10 ⁹ 0.0068 - 1.9 x 10 ¹¹ 0.36 - 130 6.8 x 10 ¹⁴ 4.4 x 10 ¹⁵ 4.4 x 10 ¹⁵ 4.4 x 10 ¹⁵ 6.2 x 10 ¹⁶ 1.5 x 10 ⁹ 0.0068 - 1.9 x 10 ¹¹ 0.36 - 130 6.8 x 10 ¹⁴ 4.4 x 10 ¹⁵ 4.5 x 10 ¹⁶ 6.2 x 10 ¹⁶ 6.2 x 10 ¹⁶ 6.2 x 10 ¹⁶ 6.3 x 10	616 (18)	Reactor centerline rear of bearing tester	Inconel X							
### MECHANICAL IRRADIATION TEST 904 (3600 SEC BURATION) 601 Reactor centerline- front face of shroud 614 Reactor centerline- front of bearing tester 615 Top, center, forward end of bearing section 7 Pump end bearing SS 302 2.8 x 10 ⁷ 2.8 x 10 ⁷ 1.4 x 10 ¹¹ 0.37 - 5.6 x 10 ¹¹ 0.26 - 4.0 1.7 x 10 ⁸ 1.7 x 10 ⁸ 1.7 x 10 ⁸ 1.7 x 10 ⁸ 1.4 x 10 ¹¹ 0.37 - 5.6 x 10 ¹¹ 0.26 - 4.0 2.0 x 10 ¹⁵ 1.3 x 10 ¹⁵ 1.3 x 10 ¹⁵ 1.3 x 10 ¹⁵ 1.5 x 10 ¹⁵ 1.9 x 10 ¹⁶ 1.7 x 10 ⁶ 8.2 x 10 ¹⁰ 0.067 - 3.7 x 10 ¹¹ 0.35 - 19 1.3 x 10 ¹⁵ 1.9 x 10 ¹⁶ 1.7 x 10 ⁶ 8.2 x 10 ¹⁷ 1.9 x 10 ¹⁰ 0.067 - 3.7 x 10 ¹¹ 0.35 - 19 1.3 x 10 ¹⁵ 1.9 x 10 ¹⁶ 1.7 x 10 ⁶ 8.2 x 10 ⁹ 0.040 - 3.6 x 10 ¹¹ 0.49 - 44 1.3 x 10 ¹⁵ 8.3 x 10 ¹⁶ 1.6 (18) Reactor centerline- feer of bearing Reactor centerline- feer of bearing tester 610 Reactor centerline- A1 6061 3.1 x 10 ⁵ 3.1 x 10 ⁵ 1.5 x 10 ⁹ 0.0068 - 1.9 x 10 ¹¹ 0.36 - 130 6.8 x 10 ¹⁴ 4.4 x 10 ¹⁴ 4.4 x 10 ¹⁵ 1.5 x 10 ⁹ 0.0068 - 1.9 x 10 ¹¹ 0.36 - 130 6.8 x 10 ¹⁴ 4.4 x 10 ¹⁵ 4.5 x 10 ¹⁶ 1.5 x 10 ⁹ 0.0068 - 1.9 x 10 ¹¹ 0.36 - 130 6.8 x 10 ¹⁴ 4.4 x 10 ¹⁵ 4.5 x 10 ¹⁶ 1.5 x 10 ⁹ 0.0068 - 1.9 x 10 ¹¹ 0.36 - 130 6.8 x 10 ¹⁴ 4.4 x 10 ¹⁵ 1.5 x 10 ⁹ 0.0068 - 1.9 x 10 ¹¹ 0.36 - 130 6.8 x 10 ¹⁴ 4.4 x 10 ¹⁵ 1.5 x 10 ⁹ 0.0068 - 1.9 x 10 ¹¹ 0.36 - 130 6.8 x 10 ¹⁴ 4.4 x 10 ¹⁵ 1.5 x 10 ⁹ 0.0068 - 1.9 x 10 ¹¹ 0.36 - 130 6.8 x 10 ¹⁴ 4.4 x 10 ¹⁵ 1.5 x 10 ⁹ 0.0068 - 1.9 x 10 ¹¹ 0.36 - 130 6.8 x 10 ¹⁴ 4.4 x 10 ¹⁵ 1.5 x 10 ⁹ 0.0068 - 1.9 x 10 ¹¹ 0.36 - 130 6.8 x 10 ¹⁴ 4.4 x 10 ¹⁵ 1.5 x 10 ⁹ 0.0068 - 1.9 x 10 ¹¹ 0.36 - 130 6.8 x 10 ¹⁴ 4.4 x 10 ¹⁵ 1.5 x 10 ⁹ 0.0068 - 1.9 x 10 ¹¹ 0.36 - 130 6.8 x 10 ¹⁴ 4.4 x 10 ¹⁵ 1.5 x 10 ⁹ 0.0068 - 1.9 x 10 ¹¹ 0.36 - 130 6.8 x 10 ¹⁴ 4.4 x 10 ¹⁵ 1.5 x 10 ⁹ 0.0068 - 1.9 x 10 ¹¹ 0.36 - 130 6.8 x 10 ¹⁴ 4.4 x 10 ¹⁵ 1.5 x 10 ⁹ 0.0068 - 1.9 x 10 ¹¹ 0.36 - 130 6.8 x 10 ¹⁴ 4.4 x 10 ¹⁵ 1.5 x 10 ⁹ 0.0068 - 1.9 x 10 ¹¹ 0.36 - 130 6.8 x 10 ¹⁴ 1.4 x 10 ¹⁵ 1.5 x 10 ⁹ 0.0068 - 1.9 x 10 ¹¹ 0.36 - 130 6.8 x 10 ¹⁴ 1.4 x 10 ¹⁵	610		Al 6061	3.1 × 10 ⁵	5.2 x 10 ⁴	1.5 × 10 ⁹	0.0068 - 1.9 x 10 ¹¹	0.36 - 130	1.1 x 10 ¹⁴	7.4 x 10 ⁻⁷
Front face of shroud 614 Reactor centerline- front of bearing tester 615 Top, center, forward end of bearing section 7 Pump end bearing 55 302 2.8 x 10 ⁷ 2.8 x 10 ⁷ 1.4 x 10 ¹¹ 1.0 - 4.9 x 10 ¹¹ 0.12 - 0.60 1.8 x 10 ¹⁵ 6.1 sector centerline- 61 Turbine end bearing 55 302 2.8 x 10 ⁷ 2.8 x 10 ⁷ 1.4 x 10 ¹¹ 0.37 - 5.6 x 10 ¹¹ 0.26 - 4.0 2.0 x 10 ¹⁵ 1.3 sector centerline- 61 Turbine end bearing 63 302 1.0 x 10 ⁷ 1.0 x 10 ⁷ 4.8 x 10 ¹⁰ 0.097 - 3.9 x 10 ¹¹ 0.20 - 8.1 1.4 x 10 ¹⁵ 4.8 sector centerline- 616 (18) Reactor centerline- Feat of bearing 618 Turbine end bearine 619 Reactor centerline- Feat of bearing 610 Reactor centerline- 610 Reactor centerline- 610 Reactor centerline- 610 Reactor centerline- 62 2.8 x 10 ⁹ 1.4 x 10 ¹² 0.37 - 5.6 x 10 ¹¹ 0.20 - 8.1 1.4 x 10 ¹⁵ 1.3 x 10 ¹⁵ 1.3 sector centerline- 63 3.1 x 10 ⁵ 1.7 x 10 ⁶ 1.7 x 10 ⁶ 1.9 x 10 ¹⁰ 0.067 - 3.7 x 10 ¹¹ 0.35 - 19 1.3 x 10 ¹⁵ 1.9 sector centerline- 64 5.2 x 10 ⁹ 0.0042 - 2.3 x 10 ¹¹ 0.49 - 44 1.3 x 10 ¹⁵ 8.3 x 10 ¹⁴ 6.2 x 10 ¹⁴ 1.4 x 10 ¹⁵ 1.1 x 10 ⁵ 1.1 x 10 ⁵ 1.5 x 10 ⁹ 0.0068 - 1.9 x 10 ¹¹ 0.36 - 130 6.8 x 10 ¹⁴ 4.4 x 10 ¹⁵ 4.4 x 10 ¹⁵ 1.5 x 10 ⁹ 0.0068 - 1.9 x 10 ¹¹ 0.36 - 130 6.8 x 10 ¹⁴ 4.4 x 10 ¹⁵ 1.5 x 10 ⁹ 0.0068 - 1.9 x 10 ¹¹ 0.36 - 130 6.8 x 10 ¹⁴ 4.4 x 10 ¹⁵ 1.5 x 10 ⁹ 0.0068 - 1.9 x 10 ¹¹ 0.36 - 130 6.8 x 10 ¹⁴ 4.4 x 10 ¹⁵ 1.5 x 10 ⁹ 0.0068 - 1.9 x 10 ¹¹ 0.36 - 130 6.8 x 10 ¹⁴ 4.4 x 10 ¹⁵ 1.5 x 10 ⁹ 0.0068 - 1.9 x 10 ¹¹ 0.36 - 130 6.8 x 10 ¹⁴ 4.4 x 10 ¹⁵ 1.5 x 10 ¹⁵				C. RADIATION ENV	VIRONMENT AND REATI RRADIATION TEST 904	ng rates for (3600 sec durat	ion)			
Tront of bearing tester 615	601		Al 6061	3.6 x 10 ⁹	3.6 x 10 ⁹	1.7 x 10 ¹³	1.2 - 1.2 x 10 ¹²	0.07 - 0.07	4.3 x 10 ¹⁵	4.8 x 10 ⁻²
Pump end bearing SS 302 2.8 x 10 ⁷ 2.8 x 10 ⁷ 1.4 x 10 ¹¹ 0.37 - 5.6 x 10 ¹¹ 0.26 - 4.0 2.0 x 10 ¹⁵ 1.3 x 10 ¹³ 13 Turbine end bearing SS 302 1.0 x 10 ⁷ 1.0 x 10 ⁷ 4.8 x 10 ¹⁰ 0.097 - 3.9 x 10 ¹¹ 0.20 - 8.1 1.4 x 10 ¹⁵ 4.8 x 10 ¹⁰ 14 Turbine end bearing SS 302 4.0 x 10 ⁶ 4.0 x 10 ⁶ 1.9 x 10 ¹⁰ 0.067 - 3.7 x 10 ¹¹ 0.35 - 19 1.3 x 10 ¹⁵ 1.9 x 10 ¹⁰ 16 Turbine end bearing SS 302 1.7 x 10 ⁶ 1.7 x 10 ⁶ 8.2 x 10 ⁹ 0.040 - 3.6 x 10 ¹¹ 0.49 - 44 1.3 x 10 ¹⁵ 8.3 x 10 ¹⁶ (18) 16 Reactor centerline- rear of bearing tester 10 Reactor centerline- A1 6061 3.1 x 10 ⁵ 3.1 x 10 ⁵ 1.5 x 10 ⁹ 0.0068 - 1.9 x 10 ¹¹ 0.36 - 130 6.8 x 10 ¹⁴ 4.4 x 10 ¹⁵ 4.5 x 10 ¹⁵ 1.5 x 10 ⁹ 0.0068 - 1.9 x 10 ¹¹ 0.36 - 130 6.8 x 10 ¹⁴ 4.4 x 10 ¹⁵ 1.5 x 10 ⁹ 0.0068 - 1.9 x 10 ¹¹ 0.36 - 130 6.8 x 10 ¹⁴ 4.4 x 10 ¹⁵ 1.5 x 10 ⁹ 0.0068 - 1.9 x 10 ¹¹ 0.36 - 130 6.8 x 10 ¹⁴ 4.4 x 10 ¹⁵ 1.5 x 10 ⁹ 0.0068 - 1.9 x 10 ¹¹ 0.36 - 130 6.8 x 10 ¹⁴ 4.4 x 10 ¹⁵ 1.5 x 10 ⁹ 0.0068 - 1.9 x 10 ¹¹ 0.36 - 130 6.8 x 10 ¹⁴ 4.4 x 10 ¹⁵ 1.5 x 10 ⁹ 0.0068 - 1.9 x 10 ¹¹ 0.36 - 130 6.8 x 10 ¹⁴ 4.4 x 10 ¹⁵ 1.5 x 10 ⁹ 0.0068 - 1.9 x 10 ¹¹ 0.36 - 130 6.8 x 10 ¹⁴ 4.4 x 10 ¹⁵ 1.5 x 10 ⁹ 0.0068 - 1.9 x 10 ¹¹ 0.36 - 130 6.8 x 10 ¹⁴ 4.4 x 10 ¹⁵ 1.5 x 10 ⁹ 0.0068 - 1.9 x 10 ¹¹ 0.36 - 130 6.8 x 10 ¹⁴ 4.4 x 10 ¹⁵ 1.5 x 10 ⁹ 0.0068 - 1.9 x 10 ¹¹ 0.36 - 130 6.8 x 10 ¹⁴ 1.5 x 10 ¹⁵ 1.5 x 10 ⁹ 0.0068 - 1.9 x 10 ¹¹ 0.36 - 130 6.8 x 10 ¹⁴ 1.5 x 10 ¹⁵ 1.5 x 10 ⁹ 0.0068 - 1.9 x 10 ¹¹ 0.36 - 130 6.8 x 10 ¹⁴ 1.5 x 10 ¹⁵ 1.5 x 10 ⁹ 0.0068 - 1.9 x 10 ¹¹ 0.36 - 130 6.8 x 10 ¹⁴ 1.5 x 10 ¹⁵ 1.5 x 10 ⁹ 0.0068 - 1.9 x 10 ¹¹ 0.36 - 130 6.8 x 10 ¹⁴ 1.5 x 10 ¹⁵ 1.5 x 10 ⁹ 0.0068 - 1.9 x 10 ¹¹ 0.36 - 130 6.8 x 10 ¹⁴ 1.5 x 10 ¹⁵ 1.5 x 10 ⁹ 0.0068 - 1.9 x 10 ¹¹ 0.36 - 130 6.8 x 10 ¹⁴ 1.5 x 10 ¹⁵ 1.5 x 10 ⁹ 0.0068 - 1.9 x 10 ¹¹ 0.36 - 130 6.8 x 10 ¹⁴ 1.5 x 10 ¹⁵	614			2.8 x 10 ⁹	2.8 x 10 ⁹	1.4 x 10 ¹³	8.8 - 9.8 x 10 ¹¹	0.063 - 0.07	3.5 x 10 ¹⁵	1.2 x 10 ⁻¹
13 Turbine end bearing S5 302 1.0 x 10 ⁷ 1.0 x 10 ⁷ 4.8 x 10 ¹⁰ 0.097 - 3.9 x 10 ¹¹ 0.20 - 8.1 1.4 x 10 ¹⁵ 4.8 : 14 Turbine end bearing S5 302 4.0 x 10 ⁶ 4.0 x 10 ⁶ 1.9 x 10 ¹⁰ 0.067 - 3.7 x 10 ¹¹ 0.35 - 19 1.3 x 10 ¹⁵ 1.9 : 16 Turbine end bearing S5 302 1.7 x 10 ⁶ 1.7 x 10 ⁶ 8.2 x 10 ⁹ 0.040 - 3.6 x 10 ¹¹ 0.49 - 44 1.3 x 10 ¹⁵ 8.3 : 616 (18) Reactor centerline- rear of bearing tester 610 Reactor centerline- A1 6061 3.1 x 10 ⁵ 3.1 x 10 ⁵ 1.5 x 10 ⁹ 0.0068 - 1.9 x 10 ¹¹ 0.36 - 130 6.8 x 10 ¹⁴ 4.4 :	615			1.7 x 10 ⁸	1.7 × 10 ⁸	8.2 x 10 ¹¹	1.0 - 4.9 x 10 ¹¹	0.12 - 0.60	1.8 x 10 ¹⁵	8.1 x 10 ⁻³
14 Turbine end bearing \$8 302 4.0 x 10 ⁶ 4.0 x 10 ⁶ 1.9 x 10 ¹⁰ 0.067 - 3.7 x 10 ¹¹ 0.35 - 19 1.3 x 10 ¹⁵ 1.9 x 10 ¹⁰ 16 Turbine end bearing \$8 302 1.7 x 10 ⁶ 1.7 x 10 ⁶ 8.2 x 10 ⁹ 0.040 - 3.6 x 10 ¹¹ 0.49 - 44 1.3 x 10 ¹⁵ 8.3 x 10 ¹⁶ (18) Reactor centerline-rear of bearing tester 610 Reactor centerline- A1 6061 3.1 x 10 ⁵ 3.1 x 10 ⁵ 1.5 x 10 ⁹ 0.0068 - 1.9 x 10 ¹¹ 0.36 - 130 6.8 x 10 ¹⁴ 4.4 x 10 ¹⁵ 1.5 x 10 ⁹ 0.0068 - 1.9 x 10 ¹¹ 0.36 - 130 6.8 x 10 ¹⁴ 4.4 x 10 ¹⁵ 1.5 x 10 ⁹ 0.0068 - 1.9 x 10 ¹¹ 0.36 - 130 6.8 x 10 ¹⁴ 4.4 x 10 ¹⁵ 1.5 x 10 ⁹ 0.0068 - 1.9 x 10 ¹¹ 0.36 - 130 6.8 x 10 ¹⁴ 1.4 x 10 ¹⁵ 1.5 x 10 ⁹ 0.0068 - 1.9 x 10 ¹¹ 0.36 - 130 6.8 x 10 ¹⁴ 1.4 x 10 ¹⁵ 1.5 x 10 ⁹ 0.0068 - 1.9 x 10 ¹¹ 0.36 - 130 6.8 x 10 ¹⁴ 1.4 x 10 ¹⁵ 1.5 x 10 ⁹ 0.0068 - 1.9 x 10 ¹¹ 0.36 - 130 6.8 x 10 ¹⁴ 1.4 x 10 ¹⁵ 1.5 x 10 ⁹ 0.0068 - 1.9 x 10 ¹¹ 0.36 - 130 6.8 x 10 ¹⁴ 1.5 x 10 ¹⁵ 1.9 x 1	7	Pump end bearing	SS 302	2.8 x 10 ⁷	2.8 x 10 ⁷	1.4 x 10 ¹¹	0.37 - 5.6 x 10 ¹¹	0,26 - 4.0	2.0 x 10 ¹⁵	1.3×10^{-3}
14 Turbine end bearing SS 302 4.0 x 10 ⁶ 4.0 x 10 ⁶ 1.9 x 10 ¹⁰ 0.067 - 3.7 x 10 ¹¹ 0.35 - 19 1.3 x 10 ¹⁵ 1.9 x 10 ¹⁰ 16 Turbine end bearing SS 302 1.7 x 10 ⁶ 1.7 x 10 ⁶ 8.2 x 10 ⁹ 0.040 - 3.6 x 10 ¹¹ 0.49 - 44 1.3 x 10 ¹⁵ 8.3 x 10 ¹⁶ (18) Reactor centerline-rear of bearing tester 610 Reactor centerline- A1 6061 3.1 x 10 ⁵ 3.1 x 10 ⁵ 1.5 x 10 ⁹ 0.0068 - 1.9 x 10 ¹¹ 0.36 - 130 6.8 x 10 ¹⁴ 4.4 x 10 ¹⁵ 1.5 x 10 ⁹ 0.0068 - 1.9 x 10 ¹¹ 0.36 - 130 6.8 x 10 ¹⁴ 4.4 x 10 ¹⁵ 1.5 x 10 ⁹ 0.0068 - 1.9 x 10 ¹¹ 0.36 - 130 6.8 x 10 ¹⁴ 4.4 x 10 ¹⁵ 1.5 x 10 ⁹ 0.0068 - 1.9 x 10 ¹¹ 0.36 - 130 6.8 x 10 ¹⁴ 4.4 x 10 ¹⁵ 1.5 x 10 ⁹ 0.0068 - 1.9 x 10 ¹¹ 0.36 - 130 6.8 x 10 ¹⁴ 4.4 x 10 ¹⁵ 1.5 x 10 ⁹ 0.0068 - 1.9 x 10 ¹¹ 0.36 - 130 6.8 x 10 ¹⁴ 4.4 x 10 ¹⁵ 1.5 x 10 ⁹ 0.0068 - 1.9 x 10 ¹¹ 0.36 - 130 6.8 x 10 ¹⁴ 0.36	13	Turbine end bearing	SS 302	1.0 x 10 ⁷	1.0 x 10 ⁷	4.8 x 10 ¹⁰	0.097 - 3.9 x 10 ¹¹	0.20 - 8.1	1.4 x 10 ¹⁵	4.8 x 10 ⁻⁴
16 Turbine end bearing 88 302 1.7 x 10 ⁶ 1.7 x 10 ⁶ 8.2 x 10 ⁹ 0.04c - 3.6 x 10 ¹¹ 0.49 - 44 1.3 x 10 ¹⁵ 8.3 x 10 ¹⁶ (18) Reactor centerlinerer of bearing tester 610 Reactor centerline- A1 6061 3.1 x 10 ⁵ 3.1 x 10 ⁵ 1.5 x 10 ⁹ 0.0068 - 1.9 x 10 ¹¹ 0.36 - 130 6.8 x 10 ¹⁴ 4.4 x 10 ¹⁵ 4.4 x 10 ¹⁵ 1.5 x 10 ⁹ 0.0068 - 1.9 x 10 ¹¹ 0.36 - 130 6.8 x 10 ¹⁴ 4.4 x 10 ¹⁵ 1.5 x 10 ⁹ 0.0068 - 1.9 x 10 ¹¹ 0.36 - 130 6.8 x 10 ¹⁴ 4.4 x 10 ¹⁵ 1.5 x 10 ⁹ 0.0068 - 1.9 x 10 ¹¹ 0.36 - 130 6.8 x 10 ¹⁴ 4.4 x 10 ¹⁵ 1.5 x 10 ⁹ 0.0068 - 1.9 x 10 ¹¹ 0.36 - 130 6.8 x 10 ¹⁴ 1.4 x 10 ¹⁵ 1.5 x 10 ⁹ 0.0068 - 1.9 x 10 ¹¹ 0.36 - 130 6.8 x 10 ¹⁴ 1.4 x 10 ¹⁵ 1.5 x 10 ⁹ 0.0068 - 1.9 x 10 ¹¹ 0.36 - 130 6.8 x 10 ¹⁴ 1.5 x 10 ¹⁵ 1.5 x 10 ⁹ 0.0068 - 1.9 x 10 ¹¹ 0.36 - 130 6.8 x 10 ¹⁴ 1.5 x 10 ¹⁵ 1.5 x 10 ⁹ 0.0068 - 1.9 x 10 ¹¹ 0.36 - 130 6.8 x 10 ¹⁴ 1.5 x 10 ¹⁵ 1.5 x 10 ⁹ 0.0068 - 1.9 x 10 ¹¹ 0.36 - 130 6.8 x 10 ¹⁴ 1.5 x 10 ¹⁵ 1.5 x 10 ⁹ 0.0068 - 1.9 x 10 ¹¹ 0.36 - 130 6.8 x 10 ¹⁴ 1.5 x 10 ¹⁵ 1.5 x 10 ⁹ 0.0068 - 1.9 x 10 ¹¹ 0.36 - 130 6.8 x 10 ¹⁴ 1.5 x 10 ¹⁵ 1.5 x 10 ⁹ 0.0068 - 1.9 x 10 ¹¹ 0.36 - 130 6.8 x 10 ¹⁴ 1.5 x 10 ¹⁵ 1.5 x 10 ⁹ 0.0068 - 1.9 x 10 ¹¹ 0.36 - 130 6.8 x 10 ¹⁴ 1.5 x 10 ¹⁵ 1.5 x 10 ⁹ 0.0068 - 1.9 x 10 ¹¹ 0.36 - 130 6.8 x 10 ¹⁴ 1.5 x 10 ¹⁵ 1.5 x 10 ⁹ 0.0068 - 1.9 x 10 ¹¹ 0.36 - 130 6.8 x 10 ¹⁴ 1.5 x 10 ¹⁵ 1.5 x 10 ⁹ 0.0068 - 1.9 x 10 ¹¹ 0.36 - 130 6.8 x 10 ¹⁴ 1.5 x 10 ¹⁵ 1.5 x 10 ⁹ 0.0068 - 1.9 x 10 ¹¹ 0.36 - 130 6.8 x 10 ¹⁴ 1.5 x 10 ¹⁵ 1.5 x 10 ⁹ 0.0068 - 1.9 x 10 ¹¹ 0.36 - 130 6.8 x 10 ¹⁴ 1.5 x 10 ¹⁵ 1.5 x 10 ⁹ 0.0068 - 1.9 x 10 ¹¹ 0.36 - 130 6.8 x 10 ¹⁴ 1.5 x 10 ¹⁵ 1.5 x 10 ⁹ 0.0068 - 1.9 x 10 ¹¹ 0.36 - 130 6.8 x 10 ¹⁴ 1.5 x 10 ¹⁵ 1.5 x 10 ⁹ 0.0068 - 1.9 x 10 ¹¹ 0.36 - 130 6.8 x 10 ¹⁴ 1.5 x 10 ¹⁵	14		***	4.0 × 10 ⁶	4-0 × 10 ⁶	1.9 x 10 ¹⁰	0.067 + 3.7 × 10 ¹¹	0.35 - 19		
616 (18) Reactor centerline-rear of bearing tester 610 Reactor centerline- Al 6061 3.1 x 10 ⁵ 1.1 x 10 ⁵ 5.3 x 10 ⁸ 0.0042 - 2.3 x 10 ¹¹ 0.79 - 430 8.3 x 10 ¹⁴ 6.2 s	= '	9	_							
rear of bearing tester 610 Reactor centerline— Al 6061 3.1 x 10 ⁵ 3.1 x 10 ⁵ 1.5 x 10 ⁹ 0.0068 - 1.9 x 10 ¹¹ 0.36 - 130 6.8 x 10 ¹⁴ 4.4.		_	•		_					
	` ,									
	610		Al 6061	3.1 x 10 ²	3.1 x 10 ⁷	1.5 x 10 ⁹	0.0068 - 1.9 x 10 ¹¹	0.36 - 130	6.8 x 10 ¹⁴	4.4 x 10 ⁻⁶

CONFIDENTIAL - RESTRICTED DATA

RN-S-0050

TABLE 5

COMPARISON OF NERVA AND ASTR ENVIRONMENT
AT THE TURBOPUMP BEARING LOCATIONS

MI'I	Test/Bearings	Maximum γ - Dose Rave (ergs/gm(c)-hr)	Maximum γ-Dose (ergs/gm(c)	Maximum γ -Flux $(\gamma/\text{cm}^2\text{sec})$	Maximum n-flux E>0.01 Mev (n/cm2sec)	Meximum nvt (n/cm²)	Maximum n/yFlux Ratio	Maximum Gamma Heating Rate btu/in -sec
901:	(i) Pump End Turbine End	2.8 x 10 ⁷ 1.6 x 10 ⁷	3.3 x L0 ⁰ 1.4 x 10 ⁰	8.7×10^{11} 7.8×10^{10}	5.6×10^{11} 3.9×10^{11}	1.8×10^{14} 1.5×10^{14}	0.64 5.0	6.7 x 10 ⁻⁴ 6.3 x 10 ⁻⁵
902:	(2) _{Pump End} Turbine End	2.8 x 10 ⁷	2.3 x 10 ⁶ 3.3 x 10 ⁵	1.4 x 10 ¹¹ 2.9 x 10 ¹⁰	5.6 x 10 ¹¹ 5.7 x 10 ¹¹	1.7 x 10 ^{1/4} 1.1 x 10 ^{1/4}	4 19	1.1 x 10 ⁻¹⁴ 1.6 x 10 ⁻⁵
903:	(3) Pump End Turbine End	2.0 x 10 ⁰	4.6 x 15 ⁵ 6.6 x 10 ⁵	1.4 x 1. ¹¹ 1.9 x 10 ¹⁰	5.6 × 10 ¹¹ 5.7 × 10 ¹¹	3.4 x 10 ¹⁴ 2.0 x 10 ¹⁴	4 19	2.2 x 10 ⁻⁴ 3.2 x 10 ⁻⁵
90h:	(4) _{Pump} End Turbine End	2.8 × 10 ⁷	2.8 x 10 ⁷ 4.0 x 10 ⁷	1.4 x 10 ¹¹ 1.9 x 10 ¹⁰	5.6 x 10 ¹¹ 3.7 x 10 ¹¹	a.e x 10 ¹⁵	4 19	1.3 x 10 ⁻³ 1.9 x 10 ⁻¹
NERVA	(5) _{Pump} End Turbine End	1.1 x 10 ⁷ 1.4 x 10 ⁸	3.7 x 10° 4.7 x 10 ⁷	5 x 10 ¹⁰ 4.3 x 10 ¹¹	1.0 x 10 ¹¹ 1.0 x 10 ¹¹	1.2 x 10 ¹⁴ 1.2 x 10 ¹⁴	2.9 0.23	5.9×10^{-5} 7.5×10^{-4}

⁽¹⁾ Integrated doses and nvt based on 327 second run deration.

Table 5

Comparison of NERVA and ASTR Environment of the Turbopump Bearing Locations

⁽²⁾ Indegrated doses and nyt based on 300 second run duration.

⁽⁵⁾ Integrated doses and nvt based on GCC second run duration.

⁽⁴⁾ Integrated doses and nvt based on 3,000 second run duration.

⁽⁵⁾ Integrated doses and nvt based on full duration run (1200 seconds).

RN-S-0050

C. MIT-903

The calculated results for MIT-903 are presented in Table 4, and are compared with NERVA values in Table 5. The difference between 902 and 903 results is the fact that the 903 test was actually run for 600 seconds whereas 902 was run for 300 seconds. The integrated doses and nvt values therefore are two times those of the 902 test.

D. MIT-904

MIT-904 is currently planned to run for 3,600 seconds (Reference 9). From a radiation level standpoint, this is the only significant planned change from 902 and 903. Radiation intensities for this test are presented in Tables 4 and 5 and are simply the 902 values increased from 300 second run duration to a 3,600 value.

Tables 3 and 5 indicate that in some cases (such as TSOV in 901) the test environment essentially mocked-up the NERVA condition, especially as regards the effects of integrated exposure. However, Table 5 shows that the bearing sets have not been exposed to integrated doses comparable to NERVA during MIT-901, 902, and 903. If a 3,600 second run can be accomplished in the 904 test, the pump-end bearings would receive double the integrated gamma dose they might receive in NERVA and the turbine-end bearings would be exposed to roughly 4% of their comparable NERVA environment for a 3,600-second 904 configuration run.

It should be remembered however, that the calculated values reported here are expected to be slightly lower than measured values if measurements could be made at the locations in question.

VI COMPARISONS WITH EXPERIMENTAL RESULTS

•		
	·	

VI. COMPARISONS WITH EXPERIMENTAL RESULTS

A number of radiation measurements have been made during the 901, 902 and 903 tests (References 2, 3, and 4). Most of these detector locations are not related to the NERVA components being tested. However, several points were located such that a comparison between these measurements and calculated values is desirable. Such a comparison of integrated gamma doses is presented in Table 6. The last column of this table shows how much the calculated value is above or below the measurement.

In interpreting these data it is important to remember that the measured results are not absolute, but actually have a fair degree of uncertainty due to counting statistics. For example, the dose point located at the reactor-end of the bearing tester was measured to be 1.6 x 10 ergs/gm(c) in test 902 for a 300 second exposure but was reported as 1.1 x 10 egs/gm(c) in test 903 when the exposure time was doubled. That is to say, the exposure time was doubled but the integrated dose was reduced at the same location in each test. This is pointed out to indicate the fact that inconsistencies exist in experimental data and is calculated-to-measured ratios which vary greatly from unity and which must be interpreted in this light, as well as considering analytical shortcomings. In the specific case mentioned previously, the calculated-to-measured ratio was 1.4 for the 300 second 902 test and 4.2 for the 600 second 903 test. It is the opinion of this report that the 902 test measurement is the more nearly correct value.

If the single point described is discarded (that is, the 4.2 measured-to-calculated value for 903), the spread of the calculations ranges from +90% to -67% of the measured values, with the exception of the dose point at the extreme end of the bearing tester where scattering effects (not included in the calculation) result in a calculated to measured ratio of 0.01. Other calculations, not reported herein, show that scattering is even more acute a problem in specific locations behind the test equipment.

In general, the calculations agree quite well with the measured results. In those areas where there are disagreements, reasons for the disagreements are obvious.

TABLE 6

COMPARISON OF CALCULATED AND EXPERIMENTAL DATA

TEST NUMBER		DETECTOR LOCATION	CALCULATED GAMMA RAY DOSE - ergs/gm(c)	MEASURED DOSE ergs/gm(c)	CALCULATED/ MEASURED RATIO
		Static flange and seal test - side nearest reactor, flange farthest from reactor center line	7.1 x 10 ⁷	8.1 x 10 ⁷	0.88
		Static flange and seal test - side nearest reactor, flange nearest reactor center line	1.6 x 10 ⁸	1.3 × 10 ⁸	1.2
901		Turbine power control valve - side nearest reactor	1.0 x 10 ⁸	8.1 x 10 ⁷	1.2
	615	Bearing tester - top, center, forward end of bearing housing	1.6 x 10 ⁷	5.4 x 10 ⁷	0.3
	616	Bearing tester - rear end of motor on reactor centerline	7.6 x 10 ¹ 4	7.0 x 10 ⁶	0.01
	601	Shroud - on reactor centerline, face nearest to reactor	3.0 x 10 ⁸	2.7 × 10 ⁸	1.1
902	614	Bearing tester - on reactor centerline, end nearest to reactor	2.3 x 10 ⁸	1.6 x 10 ⁸	1.4
	615	Bearing tester - top, center, forward end of bearing housing	1.4 x 10 ⁷	2.1 x 10 ⁷	0.67
	601	Shroud - on reactor centerline, face nearest to reactor	6.0 x 10 ⁸	3.2 x 10 ⁸	1.9
903	614	Bearing tester - on reactor centerline, end nearest to reactor	4.6 x 10 ⁸	1.1 x 10 ⁸	4.2
	615	Bearing tester - top, center, forward end of bearing housing	2.8 x 10 ⁷	5.9 x 10 ⁷	0.47

Table 6
Comparison of Calculated and Experimental Data

RN-S-0050

Measured neutron data were not obtained with Plutonium threshold or Indium detecting foils (E > 0.01 Mev and E > 0.85 Mev). For this reason comparisons of neutron fluxes have not been reported here.

:			
ı			
I			
:			

I RECEDING PAGE BLANK NOT FILMED.

REFERENCES

		:

INSCEDING PAGE DEALK NOT FILMED. RN-S-0050

REFERENCES

- 1. Analysis of the Fast Neutron and Gamma Ray Spectrum for LRP Mechanical Test 901, Aerojet-General Corp., REDN Document 2652, 30 September 1963.
- 2. NERVA Components Irradiation Program, Vol. 2: ASTR Test 1, General Dynamics NARF Document FZK-170-2, 9 December 1963.
- 3. Preliminary report to SNPO-C on ASIR 2 Irradiation Test, 16 October 1963, General Dynamics/Ft. Worth document prepared by W. M. Laney and J. D. Stanley.
- 4. Preliminary report to SNPO-C on ASTR 3 Irradiation Test, 24 October 1963, General Dynamics/Ft. Worth document prepared by C. G. Pow and J. W. Guinn.
- 5. QAD computer code developed by Los Alamos Scientific Laboratory.
- 6. Spatial and Energy Distribution of the Nuclear Field in the ASTR Radiation Effects Test Cell, General Dynamics/Ft. Worth document FZK-169, 12 August 1963.
- 7. Radiation Effects Data Book, Aerojet-General Corp., REON document 2277 to AGC-NASA SNPO, Classified CRD, 30 September 1963.
- 8. Radiation Effects Testing Data Book, Aerojet-General Corp., RECN document 2510 to AGC-NASA SNPO, classified CRD, April 1963.
- 9. <u>NERVA Radiation Effects Program Final Test Plan for MIT-904</u>, Aerojet-General Corp., Nuclear Engine Division, LRP document.